A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-327	DATE:		
CUSTOMER: JAL	MCL 15,037 DTD		
CHANGE NO: 93	MODEL: 22-3 (22M)		
TITLE: Specification Administrativ Oxygen Bottle Part Number -	ve Change (Revision to Portable - Group C Airplanes)		
ORIGIN: Convair initiated			
REASON FOR CHANGE: Specification corre	ection		
EFFECT ON WEIGHT *	EFFECT ON BALANCE *		
GUAR. WT. EMPTY OPER. WT. EMPTY O	O INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: * None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	SAL J. West		
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:		
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:			
DATE:	#ONVAIR,5.0. 6-1217 A		

Specification Administrative Change (Revision to Portable Oxygen Bottle Part Number - Group C Airplanes) Title:

Convair initiated Origin:

Reason for Change: Specification correct

Description of Change:

Page A-17, APPENDIX I-C, OXYGEN EQUIPMENT

Change the following item in the equipment list:

"1 \*Bottle, Portable From:

Oxygen and Smoke

5600-2C1A-E20B 10.0" Mask (310-liter) Scott

"1 \*Bottle, Portable To:

Oxygen and Smoke

6080-3 10.0" Mask (310-liter) Scott

Effect on Weight Empty: Effect on Balance: Effect on Performance: None

Japan Air Lines Change No. 94

Page 2 of 2

# The following shall not appear in Specification language:

- a. Machmeter Indicators and Airspeed Indicators were changed to comply with the requirements of Special Civil Air Regulation SR-450A. (Refer to Service Bulletin No. 34-12A, dated 7 December 1962.)
- b. The Engine Oil Pressure Indicator for all airplanes has been changed to SR-04A. The updating of these indicators to SR-04B is covered in Service Bulletin No. 31-1A, dated 25 May 1962.
- c. A 3924-1AG-B1-1 Turn and Bank Indicator is installed for the copilot and a 3924-1AG-C1-1 indicator installed for the pilot. These indicators are completely interchangeable and contain only minor internal differences.

Effect on Weight Empty: None Effect on Balance: None Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-327		DATE:			
CUSTOMER: JAL		MCL 15,036 DTD			
CHANGE NO: 92			MODEL: 22-3 (22M)		
	tion Administrativ Group C Airplanes)		nge (Revision to Pitot Static		
ORIGIN: Convair	initiated				
REASON FOR CHANGE:			c system to show separation of in outflow valves and cabin		
EFFECT (	ON WEIGHT *		EFF.ECT ON BALANCE *		
GUAR. WT. EMPTY	OPER. WT. EMPTY				
0	0		O INCH LB.		
EFFECT ON GUARANTEED F		one			
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL N	ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:			
		RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:		CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:					
DATE:			CONVAIR.5.D. 6-1217A		

Title: Specification Administrative Change (Revision to Pitot

Static System - Group C Airplanes)

Origin: Convair initiated

Reason for Change: To up-date pitot static system to show separa-

tion of static sources for cabin outflow valves

and cabin controller.

Description of Change:

Page 50, Para. 3.14.3.2.1 STATIC SOURCES:

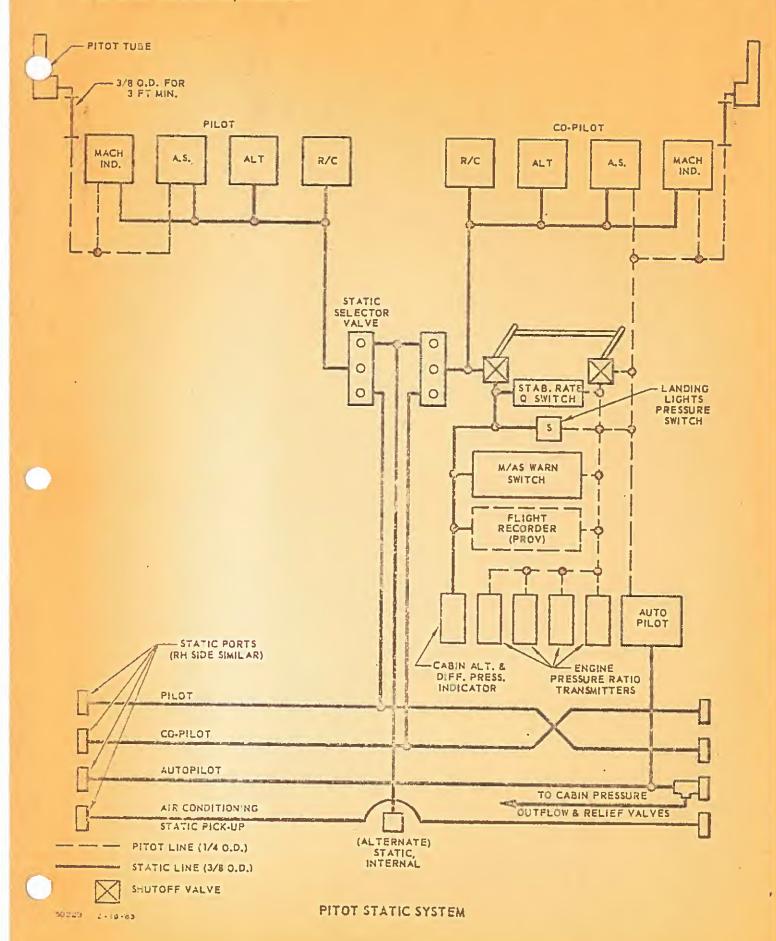
Change the last sentence to read as follows:

"A separate static source shall be provided for the autopilot system and the cabin pressure outflow valve."

Page 50a - PITOT STATIC SYSTEM - ILLUSTRATION:

Replace above illustration in the specification with attached revised illustration to show current changes.

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: 20-22-322		DATE:			
CUSTOMER: JAL		MCL 15,035 DTD			
CHANGE NO: 91			MODEL: 22	-22 (22M)	
	ation Administrate Groups A and B A			ion to Pit	ot Static
ORIGIN: Convair	init1ated				
REASON FOR CHANGE:  To up-date pitot-static system to show separation of static sources for cabin outflow valves and cabin controller.				aration and cabin	
	ON WEIGHT *		EF	FECT ON BALA	NCE *
GUAR. WT. EMPTY	OPER, WT. EMPTY				
0	0			0	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * Nor	ne			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			ags	back ENGINEERING	APPROVAL
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED	:	
RECU NON-		RECUR NON-RE	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED: CONVA		IR, A DIVISION OF	GENERAL DYNAM	ICS CORP.	
BY:					
DATE:				20	SVAIR,5.0. 6-1217.4

Title: Specification Administrative Change (Revision to Pitot

Static System - Groups A and B Airplanes)

Origin: Convair initiated

Reason for Change: To up-date pitot static system to show separation

of static sources for cabin outflow valves and

cabin controller.

Description of Change:

Page 46 Cont. 2, Para. 3.14.3.2.1 STATIC SOURCES:

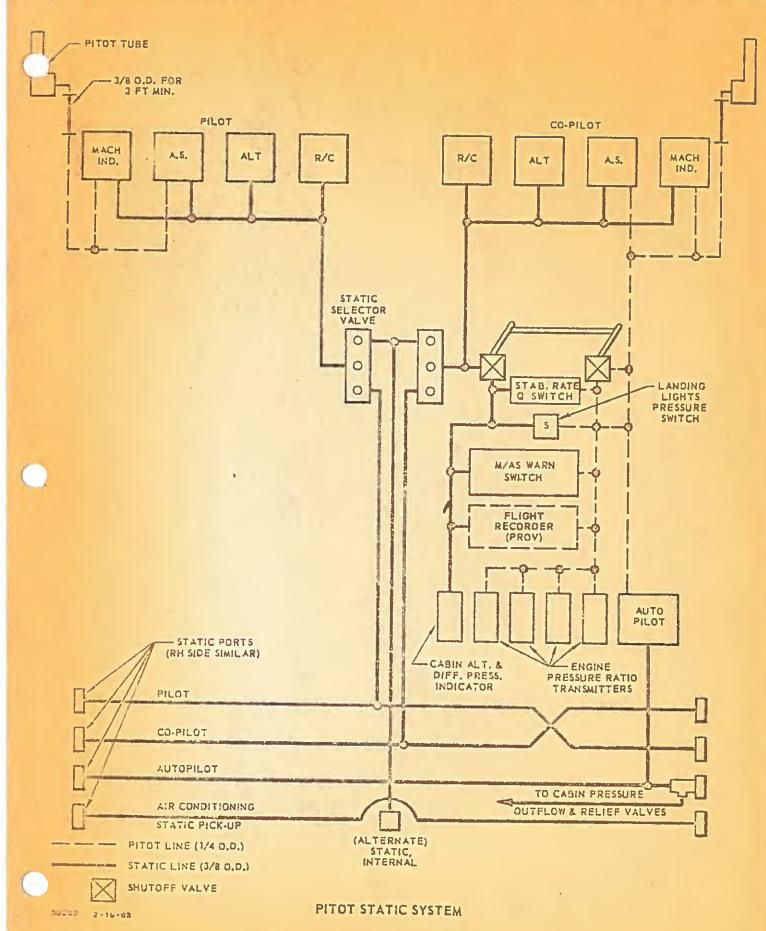
Change the last sentence to read as follows:

"A separate static source shall be provided for the autopilot system and the cabin pressure outflow valve."

Page 46a - PITOT STATIC SYSTEM - ILLUSTRATION:

Replace above illustration in the specification with attached revised illustration to show current changes.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None



A D VISIT N OF GENERAL DYNAMICS COPPORATION SAN DIE GO, CAL FORMA

SPEC NO.: ZD- 22-322		DATE:
CHANGE NO: 90		MCL 15,125 DTD
		(Convair "880 MODEL: 22-22 Ships No. 4 and No
TITLE: Relocat	ion of Doppler R/T	
ORIGIN: Custome	r request	
REASON FOR CHANGE:	As above	
EFFECT	N WEIGHT *	EFFECT ON BALANCE *
GUAR. WT. EMPTY Prop. "A" +2.0 lb Prop. "A" +2.0 lb Prop. "B" 0		Prop. "A" +1,840 In-L Prop. "B" O INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * None	
- NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF XX Change Order No. 7	FUTURE CHANGE PROPOSANGE IS DEPENDENT UPON	
LATEST DATE OF ACCEPTAN		AIRPLANES AFFECTED:
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL.
ACCEPTED-	(	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:		
DATE:	P-tr-estimated day.	200 At 1, 625 At 2

Japan Air Lines Change No. 90

Page 1 of 2

Title: Relocation of Doppler R/T Unit

Origin: Customer request

Reason for Change: As above

Description of Change:

#### PROPOSAL "A"

Page 81, Paragraph 3.17.4.3 DOPPLER NAVIGATION SYSTEM (Effective Ships 4 and 5):

Change the last sentence in the paragraph as follows:

"Installation of one receiver-transmitter (3/8 ATR Short)in accordance with ARINC Characteristic No. 540 shall be made in the hydraulic compartment."

The following shall not appear in the Specification language:

In accordance with the relocation of the Doppler R/T unit the following changes shall be accomplished:

- a. Provide mounting brackets for R/T unit on forward face of bulkhead station 1002 in the hydraulic compartment approximately in area of centerline of airplane.
- b. Provide a removable fibre glass cover to protect R/T unit from skydrol contamination.
- c. Revise hydraulic compartment shrouding to accommodate installation of R/T unit and waveguid. In addition revise waveguide installation to accommodate new unit location.
- d. Revise necessary wire harnesses and coaxial cables.
- e. Revise wiring diagram, manuals and handbooks as required.

Effect on Weight Empty: +2.0 pounds

Effect on Balance: +1,840 inch-pounds

Effect on Performance: None

Japan Air Lines Change No. 90

Page 2 of 2

# PROPOSAL "B"

The following does not affect the specification language but proposes to provide complete engineering drawings only, covering the changes described in Proposal "A". No factory work, handbooks or manual changes shall be included in this proposal.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORA 1105 SAN DIEGO, CALIFORNIA

SPEC NO. ZD- 22-322	DATE:
CUSTOMER: Japan Air Lines	MCL 15,128 DTD
CHANGE NO: 89	(Convair "8 MODEL: 22-22 Ships No. 4 and No.
TITLE: Access Door at Base of Ver	cical Stabilizer, Addition of
ORIGIN Customer request	
REASON FOR CHANGE: As above	
EFFECT ON WEIGHT	EFFECT ON BALANCE *
GUAR, WT. EMPTY OPER, WT. EMPTY	
+3.0 lb +3.0 lb	+4,410 INCH LB.
EFFECT ON GUARANTEED PERFORMANCE:	
None	
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	THE ENGINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY	
DATE:	Contract of the second of the

Title: Access Door at Base of Vertical Stabilizer, Addition of

Origin: Customer requested

Reason for Change: As above

Description of Change:

The following does not affect the specification language but proposes to accomplish the following on JAL Ships No. 4 and No. 5:

- a. Install one 14" x 18" (approximately) access door on R/H side of vertical stabilizer. Door will be located in area between Spars No. 2 and No. 3 and will be just above base of the vertical stabilizer.
- b. Door will have integral stiffeners and will be held in place with screws and nutplates.

Effect on Weight Empty: +3.0 pounds

Effect on Balance: +4,410 inch-pounds

Effect on Performance: None

SAN THE GO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:_			
CUSTOMER. Japan Air I	Lines		MCL 1	15,127	DTD 23 N	March 1962
CHANGE NO: 88					(Convair	
TITLE Oxygen Cyl:	inder Assemblies,	Subst	itutio	on of		
ORIGIN: Japan Air I	Lines requested					
REASON FOR CHANGE: As	a <b>a</b> bove					
EFFECT O	N WEIGHT *			EFFEC	T ON BALAN	ICE •
GUAR, WT. EMPTY	OPER, WT. EMPTY				•	
None	-1.0 lb				-280	INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: *  None						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL	J. 71;	short =		C, 4/3  APPROVAL
LATEST DATE OF ACCEPTA	NCE:			FECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED: CONV		CONVA	AIR, A DIN	ISION OF GE	ENERAL DYNA	MICS CORP.
DATE:						v

Title: Oxygen Cylinder Assemblies, Substitution of

Origin: Japan Air Lines requested

Reason for Change: As above

Description of Change:

Page A-16, APPENDIX I-C, OXYGEN EQUIPMENT:

Revise the below item under "Description" as follows:

From: "3 \*Cylinder, High Pressure
Oxygen (107 cu ft)
(with gages)
Zep Aero ZC268-111 126.0"

To: "3 Ø\*Cylinder, High Pressure
Oxygen (107 cu ft)
(with gages)
3 ØØ\*Cylinder, High Pressure
Oxygen (107 cu ft)

Zep Aero ZC268-111-10 126

Oxygen (107 cu ft)
(with gages)
Zep Aero

Aero (P/N to be supplied)

211

Add the following to the bottom of Page A-16:

" Applies to Ships 1 through 3. pApplies to Ships 4 and 5."

Effect on Operating Weight Empty: -1.0 pound

Effect on Balance: -280 inch-pounds

Effect on Performance: None

The following shall not appear in specification language:

This change consists of replacing the Zep Aero oxygen cylinder assemblies incorporating Robbins ROV 600 regulating valves, with Zep Aero cylinder assemblies incorporating Scott 10850-27 regulating valves on Airplanes 4 and 5.

A DIVISION OF GENERAL DYNAMICS ECOPPORATION SAN DRESO, CALLEDENIA

SPEC NO.: ZD- 22-322	DATE:
CUSTOMER: Japan Air Lines	MCL 15,124 DTD 15 March 1962
CHANGE NO: 87	MODEL: 22-22 (Convair "880")
TITLE: Doppler Antenna Access P	anel Fasteners, Replacement of
ORIGIN: Japan Air Lines requeste	d
REASON FOR CHANGE: As above	
EFFECT ON WEIGHT	EFFECT ON BALANCE
GUAF. WT. EMPTY Proposal "A" - Neg. Proposal "A" - Proposal "B" - Neg. Proposal "B" -	
EFFECT ON GUARANTEED PERFORMANCE:	None
NEGLIGIBLE CHANGES WILL BE ACCUMULATED TOTALS REFLECTED IN A FUTURE CHANGE PRO ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	UPON .
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:
ACCEPTED-BY:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
DATE	DAVE . L. THEN

Japan Air Lines Change No. 87

Page 1 of 2

Title: Doppler Antenna Access Panel Fasteners, Replacement of

Origin: Japan Air Lines requested

Reason for Change: As above

Description of Change:

#### PROPOSAL "A"

The specification language is not affected by this change which proposes to replace the present doppler antenna access door fasteners on Airplanes 4 and 5 with Camloc stressed panel fasteners No. 4R41 and No. 4S14. This will be a complete production installation and shall not include the first three JAL airplanes already delivered.

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None

Japan Air Lines Change No. 87

Page 2 of 2

# PROPOSAL "B"

The specification language is not affected by this change which proposes that complete engineering only be supplied to JAL; with no installation, manual changes, parts or service bulletins provided for.

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None



DATE: \_\_\_\_

# CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

#### COMMERCIAL CHANGE PROPOSAL

SPEC NO.: ZD- 22-322			DATE:	
CUSTOMER: Japan Air	Lines		MCL 15,123 DT	D
CHANGE NO: 86			MODEL: 22-22 Shi	conv ps #4 & #5 "8
TITLE: Inspecti	on Plates, Install	lation	of,	
ORIGIN: Customer REASON FOR CHANGE:			n holes in lower :	surface of
EFFECT	ON WEIGHT *		EFFECT ON	BALANCE *
GUAR, WT. EMPTY	OPER, WT. EMPTY			
Negligible	Negligible		Negligible	INCH LB.
EFFECT ON GUARANTEED F	PERFORMANCE: * None			
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL N		EERING APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
RECI NON-		RECUR NON-RI	T ON PRICE PER AIRPLA RING: ECURRING:	
ACCEPTED:		CONVA	IR, A DIVISION OF GENERA	L DYNAMICS CORP.
BY:				

GOSVAIR, 5, D. 6-1217 A

es . 1-4

Title: Inspection Plates, Installation of

Origin: Customer request

Reason for Change: To provide inspection holes in lower surface of

wing center section tie box.

#### Description of Change:

This change does not affect the specification language but proposes to accomplish the following on JAL Ships No. 4 and No. 5:

- a. Install two (2) six-inch diameter inspection holes in lower surface of wing center section tie box between auxiliary spars No. 1 and No. 2, and B.L 0 and 20.75.
- b. Install metal cover plates and gaskets.

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: Negligible



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:		
CUSTOMER: Japan Air	Lines		MCL 15,122	DTD	
CHANGE NO: 85			MODEL: 22-2	4 Ships #4	& #5 (Conv
TITLE: G. E. Se	rvice Bulletins or	n JAL	Engines, In	corporation	n of
ORIGIN: Customer	request		**		
REASON FOR CHANGE:	As above				
EFFECT	N WEIGHT *		EFF	ECT ON BALAN	ICE *
GUAR, WT. EMPTY	OPER. WT. EMPTY				
+17.0 lb	+17.0 lb			+13,175	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * Nor.	ne			
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	M. L C 2/15/62	ENGINEERING	APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED		
RECU NON-I		RECUR NON-R	RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED:  BY:		CONVA	AIR, A DIVISION OF	GENERAL DYNAM	AICS CORP.

Title: G.E. Service Bulletins on JAL Engines, Incorporation of

Origin: Customer request

Reason for Change: As above

Description of Change:

The specification language is not affected by this change which proposes to accomplish the following:

\*Incorporation of the following G.E. service bulletins on JAL engines for Ships No. 4 and No. 5, during the manufacturing cycle of the airplanes:

72-39 Welding of Oil Tank CSD Bulkhead 72-41 Addition of Stiffeners and Bearing Plate to Oil Tanks

73-5 Modification of CIT Sensor

75-2 CDT Controlled Anti-Icing System

75-3 Clamping of Anti-Icing Signal Line 78-6 Retrofit of Oil Filter Body, Cap and Chip Detector

\*Effective Ships 4 and 5.

Effect on Weight Empty: +17.0 pounds

Effect on Balance: +13,175 inch-pounds

Effect on Performance: None

A second

#### CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, GALIFORNIA

SPEC NO.: ZD-22-322		DATE			
CUSTOMER Japan Air Lines		MCL	15,115	DTD	
CHANGE NO: 83		MOD	EL: 22-22	(Conv	vair "880")
TITLE: Change Crew Oxygen Masks from Convair Furnished to Customer Furnished (Ships 4 and on)					Customer
ORIGIN: JAL Letter	r, SANQA-X-285, d	dated 29	May 1961		
REASON FOR CHANGE: Cui	stomer request				
EFFECT ON	WEIGHT *		EFFE	T ON BALA	NCE *
GUAR, WT. EMPTY	PER. WT. EMPTY				
Negligible	Negligible		Neglig	ible	INCH LB.
EFFECT ON GUARANTEED PER	EFFECT ON GUARANTEED PERFORMANCE: *  None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			le E	UN 18	72/61 G APPROVAL
LATEST DATE OF ACCEPTANCE:		AIRPLANES	RPLANES AFFECTED:		
		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED: CONV.		CONVAIR, A D	IVISION OF GEI	NERAL DYNA	MICS CORP.
BY:					

Title: Change Crew Oxygen Masks From Convair Furnished to Customer

Furnished (Ships 4 and on)

Origin: JAL Letter, SANQA-X-285, dated 29 May 1961

Reason for Change: Customer request

Description of Change:

Page A-1, APPENDIX I-A, CUSTOMER FURNISHED - CONVAIR INSTALLED

Add the following to the equipment list:

"\*\*5 Mask, Supplemental Oxygen
(with Carbon Mike RS-84) AF

ARO 23808-100A-M

\*\*5 Hoses, Oxygen ARO 23808-120K"

Add the following to bottom of page:

"\*\*Effective Ships 4 and on."

Page A-16, APPENDIX I-C, CONVAIR FURNISHED - CONVAIR INSTALLED
OXYGEN EQUIPMENT

Change the following items in the equipment list:

From: "5 Mask, Supplemental Oxygen Sierra 358-158"

To: "\*\*5 Mask, Supplemental Oxygen Sierra 358-158"

Add the following to bottom of page:

"\*\*Effective Ships 1 through 3."

Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None

A DISTRIBUTE OF GENERAL DYNAMICS CORPORATION SAN LIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE			
CUSTOMER: Japan Air Lines	MCL 15,034 DTD			
CHANGE NO. 82A	MODEL: 22-22 (Convair "880")			
Specification Administrat Radar Indicator Part Numb	cive Change (Revision to Weather ber)			
ORIGIN: Convair initiated				
REASON FOR CHANGE: Specification correction, to specify a later improved weather radar indicator including an erase feature, and revision to CCP No. 82.				
EFFECT ON WEIGHT *	EFFECT ON BALANCE			
O O O	O INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *  None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROP ACCEPTANCE OF THIS CHANGE IS DEPENDENT UP PRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	OSAL M. I.			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS.	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED: BY:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			

Title: Specification Administrative Change (Revision to Weather

Radar Indicator Part Number)

Origin: Convair initiated

Reason for Change: Specification correction, to specify a later

improved weather radar indicator including an

erase feature, and revision to CCP No. 82.

Description of Change:

Page 80, Paragraph 3.17.4.1 WEATHER PENETRATION AIRBORNE RADAR:

Add the following to end of paragraph:

"The indicator shall be equipped with a bright cathode ray tube."

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT

Change the following item under "Weather Radar":

From: "1 Indicator RCA M1-23613-3

To; "1 Indicator RCA M1-23613-5"

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF GENERAL EYNAMICS CORPORATION SAN DIFFO, CALIFORNIA

SPEC NO.: ZD-22-322	DATE:
CUSTOMER: Japan Air Lines	MCL 15,118 DTD
CHANGE NO: 81	MODEL: 22-22 (Convair "880")
TITLE: EPR Indicator, Revision to	
ORIGIN: Customer verbal request	
REASON FOR CHANGE: As above	• -
EFFECT ON WEIGHT *	EFFECT ON BALANCE *
GUAR, WT. EMPTY OPER, WT. EMPTY	
+4.0 lb +4.0 lb	+816 INCH LB.
EFFECT ON GUARANTEED PERFORMANCE: * None	
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	SAL MHady
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	
DATE *	

Title: EPR Indicator, Revision to

Origin: Customer verbal request

Reason for Change: As above

Description of Change:

Page A-5, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Change the below equipment under "Engine Instruments - General" as follows:

114 From: Indicator Pressure Ratio Kollsman A31537-10-001" 11 #4 To: Indicator Pressure Ratio Kollsman A31537-10-001

\*\*4 Indicator Pressure Ratio Kollsman C29187-10-004"

Add the following to bottom of page:

" \*Effective Ships 1, 2 and 3.

\*\*Effective Ships 4 and on."

Effect on Weight Empty: Effect on Balance: +4.0 pounds +816 inch-pounds

Effect on Performance: None



ALL OF TO, CALIFORNIA

SPEC NO. ZD- 22-322		DATE:		
CUSTOMER: Japan Air Lines		MCL 15,110 DTD		
CHANGE NO: 80		MODEL 22-22 (Convai	Lr "880")	
TITLE CFE Single Doppler,	Installation	of		
ORIGIN. JAL, TWX TYOQA/CPG/	130800, dated	13 April 1961		
REASON FOR CHANGE: Customer	request			
EFFECT ON WEIGHT *		EFFECT ON BALANCE		
GUAR, WT. EMPTY OPER, WT. EM	PTY			
+101.0 1b +101	.0 lb	+66,293	INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE:	None			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		Attaty 8/26, ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTANCE:	AIRPLA	ANES AFFECTED:		
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE:  RECURRING		
ACCEPTED.		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY.				
DATE				

Page 1 of 4

Title: CFE Single Doppler, Installation of

Origin: JAL TWX TYOQA/CPG/130800, dated 13 April 1961

Reason for Change: Customer request

Description of Change:

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS - ILLUSTRATION

Revise above illustration to show ship effectivity 1, 2 and 3.

Page 44a-1, PILOT'S AND COPILOT'S INSTRUMENT PANELS - ILLUSTRATION

Add the above illustration which shall show ship effectivity 4 and 5, and the installation of Doppler indicator.

Page 44b, PILOT'S AND COPILOT'S PEDESTAL - ILLUSTRATION

Revise above illustration to show ship effectivity 1, 2 and 3.

Page 44b-1, PILOT'S AND COPILOT'S ILLUSTRATION

Add the above illustration which shall show ship effectivity 4 and 5, and the installation of a Doppler computer control panel.

Page 70, Paragraph 3.17.1 EQUIPMENT:

Change the below items under "Provisions for the following systems shall be made":

From: "Doppler Navigation System (Single)
Doppler Computer Mark I (Single)"

To: "\*Doppler Navigation System (Single)
\*Doppler Computer Mark I (Single)"

Add the following items under the description list:

"\*\*Doppler Navigation System (Single) (Customer Furnished)
\*\*Doppler Computer Mark I (Single) (Customer Furnished)

Add the following to bottom of page:

"\*Effective Ships 1, 2 and 3.

\*\*Effective Ships 4 and 5."

Page 71, Paragraph 3.17.1.1 CONTROL PANELS:

Add the following items under the description list:

"\*\*Doppler Tracker (Customer Furnished)

\*\*Doppler Navigation Computer (Customer Furnished)"

Page 2 of 4

Change the below items under "Provisions for the following control panels shall be made":

From: "Doppler Sensor

Doppler Navigation Computer"

To: "\*Doppler Sensor

\*Doppler Navigation Computer"

Add the following to bottom of page:

"\*Effective Ships 1, 2 and 3

\*\*Effective Ships 4 and 5"

Page 80a, ANTENNA ARRANGEMENT - ILLUSTRATION

Revise above illustration to show ship effectivity 1, 2 and 3.

Page 80a-1, ANTENNA ARRANGEMENT - ILLUSTRATION

Add the above illustration which shall show ship effectivity 4 and 5, and the installation of a Doppler antenna.

Page 81, Paragraph 3.17.4.3 DOPPLER NAVIGATION SYSTEM (PROVISIONS):

Change the above paragraph title to read as follows:

"DOPPLER NAVIGATION SYSTEM (PROVISIONS) (Effective Ships 1, 2 and 3)"

Add the following new paragraphs:

- "3.17.4.3 DOPPLER NAVIGATION SYSTEM (Effective Ships 4 and 5): Installation of one tracker unit (1/2 ATR Long) in accordance with ARINC Characteristics No. 540 shall be made in the radio rack. Installation of one transmitter/receiver (3/8 ATR Short) in accordance with ARINC Characteristics No. 540 shall be made in the fuselage as close as possible to the antenna."
- "3.17.4.3.1 CONTROL (Effective Ships 4 and 5): Installation of one Doppler radar control panel shall be made in the pilot's and copilot's forward pedestal."
- "3.17.4.3.2 INDICATOR (Effective Ships 4 and 5): Installation of a Doppler radar indicator (single system) shall be made in the pilot's and copilot's instrument panels."

Page 3 of 4

- "3.17.4.3.3 ANTENNA (Effective Ships 4 and 5): A Doppler antenna assembly consisting of an antenna wave guide, adapter and wiring shall be installed on the left hand wing outboard of the main landing gear wheel well. This antenna assembly shall also permit operation of a second Doppler system"
- Page 81, Paragraph 3.17.4.4 DOPPLER NAVIGATION COMPUTER (PROVISIONS):

Change the above paragraph title to read as follows:

"DOPPLER NAVIGATION COMPUTER (PROVISIONS) (Effective Ships 1, 2 and 3):"

Add the following new paragraphs:

- "3.17.4.4 DOPPLER NAVIGATION COMPUTER (Effective Ships 4 and 5):
  Installation of one (1/4 ATR Short) computer unit in
  accordance with ARINC Characteristics No. 543 shall be
  made in the radio rack."
- "3.17.4.4.1 CONTROL (Effective Ships 4 and 5): Installation of a Doppler navigator computer controller shall be made in the pilot's and copilot's forward pedestal."

Page A-1, APPENDIX I-A, CUSTOMER FURNISHED - CONVAIR INSTALLED:

Add the following items to the equipment list under a new heading as follows:

	ELECTRONICS		CEB
DOP	PLER NAVIGATION	22	- 03537
di	Receiver/Transmitter	Bendix	-// RTA-12A
\$1 \$1 \$1 \$1 \$1 \$1	Tracker Unit	Bendix	-/3 DRA-12A
øl	Navigation Computer	Bendix	-15 CPA-24A
ø1	Computer Controller	Bendix	-9 CNA-24B-1
ø1	Control Unit	Bendix	-7 CNA-12B-1
ø1	Adapter Assembly	Bendix	-5 ADA-12C-2
ø1	Antenna (Dual)	Bendix	-3ANA-12C
ø2	Indicator (Clamp Mounted)	Bendix	-/INA-12B-1

Add the following to the bottom of page:

<sup>&</sup>quot;øEffective Ships 4 and 5."

Japan Air Lines Change No. 80

Page 4 of 4

- (A) One copy Page 44a and 44a-1 PILOT'S AND COPILOT'S INSTRUMENT PANELS Enclosures:
  - (B) One copy Page 44b and 44b-1 - PILOT'S AND COPILIT'S PEDESTAL
  - (C) One copy Page 80a and 80a-1 ANTENNA ARRANGEMENT

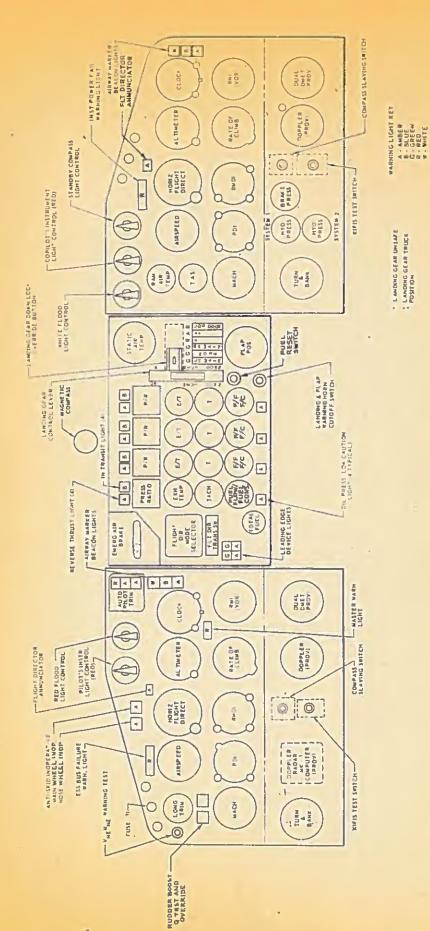
The following shall not appear in the specification language: Handbooks and manuals will not be revised to reflect the above changes.

Effect on Weight Empty: +101.0 pounds

Effect on Balance: +66,293 inch-pounds

Effect on Performance: None

3/18 3/31 6/32A

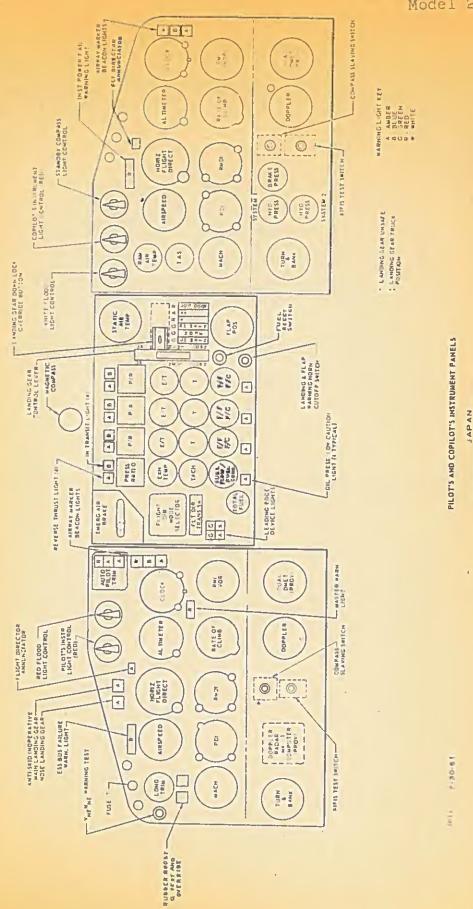


PILOT'S AND COPILOT'S INSTRUMENT PANELS

GEFECTIVE SHIPS 1. 2 AND \$1

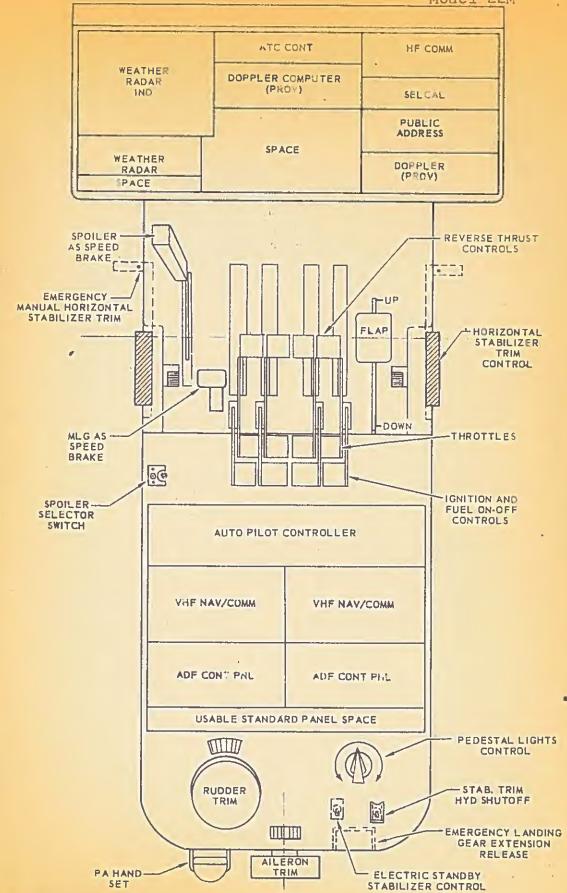
(c) 11 6 - 24 - 61

(EFFECTIVE SHIPS 4 AND SI



3/18

Page 44b
Report No. ZD-22-322
Model 22M

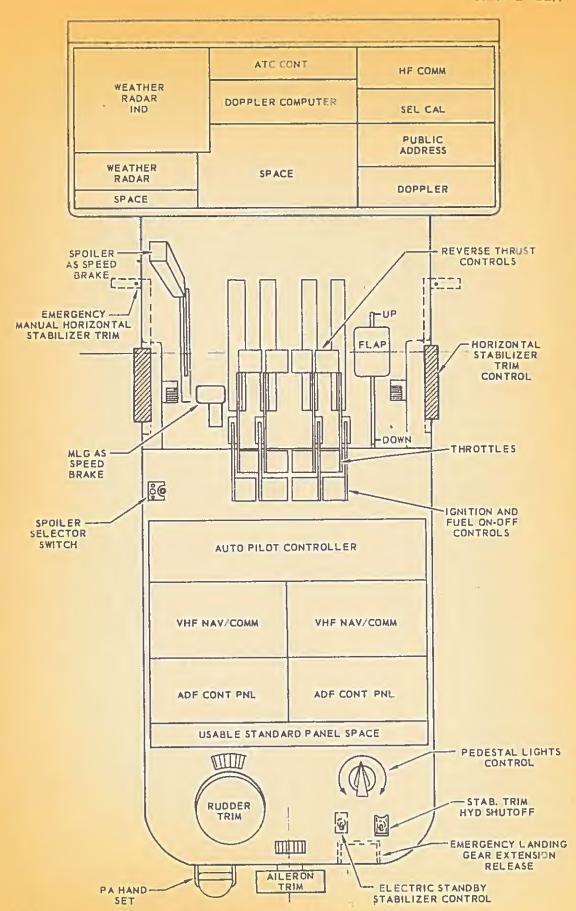


5021F 7-21-60

PILOT'S AND CO-PILOT'S PEDESTAL

JAPAN (EFFECTIVE SHIPS 1, 2 AND 3)

5021F 3.5 - 1



PILOT'S AND CO-PILOT'S PEDESTAL

JAPAN (EFFECTIVE SHIPS 4 AND 5) 1 -1

## CONVAIR

SHEC NO. ZD- 22-322		DATE	DATE			
CUSTOMER: Japan Air Lines		MCL 15,119 DTD				
CHANGE NO- 79	·	MODEL: 22-22 (Convair "880	")			
TITLE: ATC Marker	Beacon Panel, Re	vision to				
ORIGIN: Customer r	equest					
REASON FOR CHANGE: A	s above					
EFFECT (	ON WEIGHT *	EFFECT ON BALANCE *				
GUAR, WT. EMPTY	OPER. WT. EMPTY					
0	0	O INCH I	В,			
EFFECT ON GUARANTEED F	PERFORMANCE: * None	е				
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPO ANGE IS DEPENDENT UPO	DN PORTLY 84/6, MA	i]N			
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES AFFECTED:				
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.				
BŤ.		- 0-01				

Title: ATC Marker Beacon Panel, Revision to

Origin: Customer request

Reason for Change: As above

Description of Change:

Page A-11, Appendix I-C, ELECTRONIC EQUIPMENT:

From:	"I Dual ATC Transponder Beacon Control Panel	<b>G</b> ables	<b>a-</b> 565"
To:	"*1 Dual ATC Transponder Beacon Control Panel **1 Dual ATC Transponder Beacon	Gables	<b>a</b> -565
	Control Panel	Gables	G-565A"

Add the following to the bottom of page:

" \*Effective Ships 1, 2 and 3.

\*\*Effective Ships 4 and on."

The following not to appear in the specification language:

Provisions shall be made for on-off-dimming of the marker beacon panel light from the pedestal control switch.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: 0

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

#### COMMERCIAL CHANGE PROPOSAL

SPEC NO .: ZD- 22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,117 DTD			
CHANGE NO: 78			MODEL: 22-22	(Convair	"880")	
TITLE: External Po	ower Circuit Chang	е				
ORIGIN: Customer re	equest					
REASON FOR CHANGE:	As above					
EFFECT O	ON WEIGHT *		EFFEC	T ON BALANCI	E *	
GUAR. WT. EMPTY	OPER. WT. EMPTY					
+5.0 lb +5.0 lb			+1,8	325	INCH LB.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None					
NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL N		Ly 72461		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
RECL NON-I		RECUR NON-R	CT ON PRICE PER AL RRING: ECURRING:			
ACCEPTED:  BY:		CONVA	AIR, A DIVISION OF GEI		S CORP.	
DATE:				12.4.4	A 1 1, 34 2 341 2 1 1 4	

Title: External Power Circuit Change

Origin: Customer request

Reason for Change: As above

Description of Change:

Page 56, Paragraph 3.16.5.10 WIRE IDENTIFICATION:

Delete the period after the first sentence and add the following:

"except, that wire identification code numbers in the external power receptacles and junction box circuits will not conform to Specification 0-9001."

Page 66, Paragraph 3.16.10.3 EXTERNAL POWER:

Add the following after the first sentence:

"28V DC power, derived from aircraft 115v ac system shall be supplied to ground power unit through external power receptacle to control ground power main contactor."

The following shall not appear in the specification language:

Handbooks and manuals will not be revised to reflect the above change.

Effect on Weight Empty: +5.0 pounds

Effect on Balance: +1,825 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:
CUSTOMER: Japan Air Lines	MCL 15,033 DTD
CHANGE NO: 77A .	MODEL: 22-22 (Convair "880")
TITLE: Specification Administrative Hook)	e Change (Revision to Lavatory Coat
ORIGIN: Convair initiated  REASON FOR CHANGE: To clarify the int to CCP No. 77.	ent of the specification and revision
EFFECT ON WEIGHT *	EFFECT ON BALANCE *
GUAR. WT. EMPTY OPER. WT. EMPTY	O INCH LB.
0 0	O INCITED.
EFFECT ON GUARANTEED PERFORMANCE: *	ne
*-NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROPACCEPTANCE OF THIS CHANGE IS DEPENDENT UPPRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	POSAL PON ES:
	ENGINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:	
DATE:	5047A(14,2)_1(217

Title: Specification Administrative Change (Revision to Lavatory

Coat Hook)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification and

revision to CCP No. 77

Description of Change:

Page 86, Paragraph 3.19.2.2.3 MISCELLANEIOUS LAVATORY EQUIPMENT:

Revise below item as follows:

From: "One coat hook (folding-type on lavatory door)"

To: "One coat hook"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A DAVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO .: ZD- 22-322		!	DATE:			
CUSTOMER: Japan Air	CUSTOMER: Japan Air Lines		MCL 15,113A DTD			
CHANGE NO: 76A		MODEL: 22-22 (Convair "880"			r "880")	
TITLE: Bleed Duct Installati	Pressure System Son of (For Airplan	Transm nes 4	itter Mount and 5)	and Brack	et,	
ORIGIN: JAL Letter	SANQA-X-273 dated	d 23 M	ay 1961.			
REASON FOR CHANGE:	Customer requested	d and	revision to	CCP No. 7	6.	
EFFECT O	N WEIGHT *		EFF	ECT ON BALAN	ICE *	
GUAR, WT. EMPTY	OPER. WT. EMPTY					
+1.0 lb	+1.0 lb	+645 INCH		INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: * None					
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL Ni	m. ec.	3/1/62 ENGINEERING	APPROVAL	
LATEST DATE OF ACCEPTA		AIRPLA	ANES AFFECTED	D:		
SPECIAL PROVISIONS;		RECUR NON-RE	T ON PRICE PER RING: ECURRING:			
ACCEPTED:  BY-		CONVA	IR, A DIVISION OF	GENERAL DYNA	AICS CORP.	

A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-327  CUSTOMER: JAPAN AIR LINES		DATE:					
		MCL 15,038 DTD 5 June 1963					
CHANGE NO: 94		MODEL: 22-3 Ship #46 (Convair 8					
TITLE: Specificat Various In	ion Administrative struments)	Chan	ge (Part Numb	er Revis	ion to		
ORIGIN: Convair in	itiated						
REASON FOR CHANGE: TO	o reflect actual in the airplane.	Instru	ment part num	bers ins	stalled		
	ON WEIGHT *		EFFE	CT ON BALA	NCE *		
GUAR. WT. EMPTY	OPER, WT. EMPTY						
0	0			0	INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: * None						
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	-65-	NGINEERING	G APPROVAL		
LATEST DATE OF ACCEPTA	NCE:	AIRPLA	ANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:					
RECURRING:		WKKING:					
ACCEPTED:		CONVAI	R, A DIVISION OF GE	NERAL DYNA	MICS CORP.		
BY:							
DATE:				â	ONVAIR,5 D. 6-12174		

Ť.,

Japan Air Lines Change No. 94

Page 1 of 2

Title: Specification Administrative Change (Part Number Revision to

Various Instruments)

Origin: Convair initiated

Reason for Change: To reflect actual instrument part numbers installed

in the airplane.

Description of Change:

Page A-4, APPENDIX I-C - INSTRUMENTS AND RELATED EQUIPMENT

Under - Flight Instruments

Change: "2 Indicator, Turn and Bank Pioneer-Central 3924-1AG-B1-1

To: ø2 Indicator, Turn and Bank Pioneer-

Central 3924-1AG-C1-1

2 Indicator, Turn and Bank

Pioneer Gentral 3924-1AG-B1-1"

Add the following note to bottom of page:

"NOTE: ø = Installed in 22-3 Ship #4 only."

#### Page A-5, APPENDIX I-C - INSTRUMENTS AND RELATED EQUIPMENT

## Under - Kollsman Integrated Flight Instrument System

B34627-10-031 115 Indicator, Airspeed Kollsman Change: B34627-10-046 Kollsman ø2 Indicator, Airspeed To: B34627-10-031" Indicator, Airspeed Kollsman 112 A29247-10-006 Indicator, Machmeter Kollsman Change:

To: \$\\ \psi^2 \quad \text{Indicator, Machmeter Kollsman A29247-10-014 } \\ \text{Indicator, Machmeter Kollsman A29247-10-006"} \end{align\*}

Add the following note to bottom of page:

"NOTE:  $\phi = Installed in 22-3 Ship #4 only."$ 

#### Page A-7, APPENDIX I- C - INSTRUMENTS AND RELATED EQUIPMENT

Under - Engine Instruments - Oil System

Change: "4 Indicator, Oil Pressure U. S. Gauge SR-04B

To: 4 Indicator, Oil Pressure U. S. Gauge SR-04A"

Bleed Duct Pressure System Transmitter Mount and Bracket, Title:

Installation of (For Airplanes 4 and 5)

JAL Letter SANQA-X-273, dated 23 May 1961

Reason for Change: Customer request and revision to CCP No. 76.

Description of Change:

The specification language is not affected by this change which proposes to install provisions for a bleed duct pressure system transmitter consisting of the following items:

A support housing and bracket.

b. A Robinson shock-mount.

Effect on Weight Empty:

+1.0 pound +645 inch-pound Effect on Balance:

Effect on Performance: None

0

A DIVISION OF GENERAL DYNAMICS CORPORATION AND DIEGO, CALIFORNIA

## COMMERCIAL CHANGE PROPOSAL

SPEC NO.: ZD- 22-322	DATE:			
CUSTOMER: Japan Air Lines	MCL 15,113 DTD 24 May 1961			
CHANGE NO: 76	MODEL:22-22 (Convair "880")			
TITLE: Bleed Duct Pressure Indica 4 and 5)	tor, Installation of (For Airplanes			
ORIGIN: JAL Letter SANQA-X-273, da	ted 23 May 1961			
REASON FOR CHANGE: Customer request				
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR. WT. EMPTY OPER. WT. EMPTY				
+6.0 lb +6.0 lb	+3,289 INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *				
None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROPORTION ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPPER ACCEPTANCE OF THE FOLLOWING CHANGE:	OSAL ON S: Ready 7/21/61 (mylalla)			
	ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:			
	RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:				
DATE:	CONVAIR,5,0, 64217A			

4.

Title: Bleed Duct Pressure Indicator, Installation of (For Airplanes 4 and 5)

Origin: JAL Letter SANQA-X-273, dated 23 May 1961

Reason for Change: Customer request

Description of Change:

Page 45, Paragraph 3.14.1.4 FLIGHT ENGINEER'S INSTRUMENTS:

Add the following item to the instrument list:

"\*\*One Bleed Duct Pressure Indicator"

Add the following to the bottom of Page 45:

"\*\*Effective Ships 4 and 5"

Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Add the following item under "PRESSURIZATION, HEATING AND AIR CONDITIONING":

"#1 Indicator, Bleed Duct Pressure U. S. Gauge SR-05A"

Add the following to the bottom of Page A-7:

"\*Effective Ships 4 and 5"

Effect on Weight Empty: +6.0 pounds Effect on Balance: +3,289 inch-

+3,289 inch-pounds

Effect on Performance: None

SAN DIE DO, CALIFORNIA

SPEC NO. ZD- 22-322		DATE:
CUSTOMER Japan Air	Lines	MCL 15,116 DTD 21 June 1961
CHANGE NO. 75		MODEL: 22-22 (Convair "880")
TITLE Dimming Featlation of	ture in Forward L.	H. Lavatory Light Circuit, Instal-
GRIGIN. Japan Air L	ines Letter SANQA-	-X-334 dated 19 June 1961
REASON FOR CHANGE: 0	ustomer requested	
EFFECT	DN WEIGHT *	EFFECT ON BALANCE *
GUAH. WT. EMPTY	OPER. WT. EMPTY	
Negligible Negligible		NegligibleNCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * None	•
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOSANGE IS DEPENDENT UPON	ENGINEERING APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:
ACCEPTED-		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.

Page 1 of 2

Title: Dimming Feature in Forward L.H. Lavatory Light Circuit,
Installation of

Origin: Japan Air Lines Letter SANQA-X-334 dated 19 June 1961.

Reason for Change: Customer request

Description of Change:

Page 63 (Cont), Paragraph 3.16.8.3.6 SIGNS:

Revise the paragraph into two paragraphs, as follows, to indicate airplane effectivity:

- "3.16.8.3.6 (Effective on Airplanes 1 through 3): SIGNS Illuminated and legible "NO SMOKING" and "FASTEN SEAT BELT" signs shall be provided in the passenger compartment and a "RETURN TO CABIN" sign shall be provided in each lavatory. A switch for the "NO SMOKING" and "FASTEN SEAT BELT" shall be located in the pilot compartment. The switch shall be a three-position "OFF" center type. The forward position shall be labeled "NO SMOKING AND FASTEN SEAT BELT". The aft position shall be labeled "FASTEN SEAT BELT". Electrically lighted signs shall be plainly legible to passengers under all conditions of ambient light encountered.
  Illuminated lavatory "OCCUPIED-VACANT" signs shall be installed for each lavatory so as to be plainly visible to passengers. The signs shall be energized when the lavatory door lock is actuated. The locked position shall indicate "OCCUPIED" and the unlocked position shall indicate "VACANT". Reference 3.2.5 for bilingual, trilingual description.
- 3.16.8.3.6 SIGNS (Effective on Airplanes 4 and 5): Illuminated and legible "NO SMOKING" and "FASTEN SEAT BELT" signs shall be provided in the passenger compartment and a "RETURN TO CABIN" sign shall be provided in each lavatory. A switch for the "NO SMOKING" and "FASTEN SEAT BELT" shall be located in the pilot compartment. The switch shall be a three-position "OFF" center type. The forward position shall be labeled "NO SMOKING AND FASTEN SEAT BELT". The aft position shall be labeled "FASTEN SEAT BELT". Electrically lighted signs shall be plainly legible to passengers under all conditions of ambient light encountered. Illuminated lavatory "OCCUPIED-VACANT" signs shall be

GENERAL DYNAMICS/CONVAIR

Japan Air Lines Change No. 75

Page 2 of 2

installed for each lavatory so as to be plainly visible to passengers. The signs shall be energized when the lavatory door lock is actuated. The "OCCUPIED-VACANT" light circuit in the forward left hand lavatory shall be provided with a dimming feature controlled by a protected "BRIGHT-DIM" switch located in the forward left hand hat rack. The locked position shall indicate "OC-CUPIED" and the unlocked position shall indicate "VACANT". Reference 3.2.5 for bilingual, trilingual description.

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO. ZD- 22-322	DATE:
CUSTOMER: Japan Air Lines	MCL 15,112 DTD 23 May 1961
CHANGE NO: 74	MODEL: 22-22 (Convair "880")
TITLE Loadable Cargo Size Cha	arts, Furnishing of
ORIGIN: JAL AVO, dated 26 April	1 1961
REASON FOR CHANGE: Customer re	quest
EFFECT ON WEIGHT *	EFFECT ON BALANCE *
O OPER. WT. EMPTY O	O INCH I B
EFFECT ON GUARANTEED PERFORMANCE: *	None
NEGLIGIBLE CHANGES WILL BE ACCUMULATE TOTALS REFLECTED IN A FUTURE CHANGE ACCEPTANCE OF THIS CHANGE IS DEPENDENT PRIOR ACCEPTANCE OF THE FOLLOWING COMMON CO	ENT UPON (1/4)
	ENGINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.
ВҮ	

#### GENERAL DYNAMICS/CONVAIR

Title: Loadable Cargo Size Charts, Furnishing of

Origin: JAL AVO, dated 26 April 1961

Reason for Change: Customer request

Description of Change:

The specification language is not affected by this change which proposes to furnish data charts listing permissible cargo sizes that may be loaded into both the forward and aft cargo compartments.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO .: ZD- 22-322		0	DATE:		
	JAPAN AIR LINES MCL 15,032 DTD				
CHANGE NO: 73				"880")	
	Lon Administrative	Chang	e (Revision	to Paragr	aph
ORIGIN: Convair in:	lt1ated				
REASON FOR CHANGE: S	pecification clari	ficati	Lon.		
FFFFCT	ON WEIGHT *		EFFE	CT ON BALAN	NCE *
GUAR, WT. EMPTY	OPER. WT. EMPTY		<del></del>		
0	0		v	0	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *				
	None			0	
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	All	MATE (	97/, SAPPROVAL
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED:		CONVA	IR, A DIVISION OF G	ENERAL DYNA	MICS CORP.
BY:					
DATE:				-	(0% VAII) (5,5,5,64217

Japan Air Lines Change No. 73

CONVAIR: SD

Title: Specification Administrative Change (Revision to Paragraph

3.2.2.1)

Origin: Convair initiated

Reason for Change: Specification clarification.

Description of Change:

Page 9, Paragraph 3.2.2.1 COLOR CODING:

Revise the paragraph as follows:

Prom: "COLOR CODING: All sections of plumbing shall be color coded and banded in accordance with Military Standards and, in addition, marking shall indicate the direction of flow and the purpose of the line. The part number of each tubing assembly shall be identified. Color bands and part number identifications shall be plastic tape in accordance with Standard AND 10375."

To:

"COLOR CODING: All sections of plumbing, except engine and within fuel tanks, shall be color coded and banded. All plumbing shall indicate the direction of flow and the purpose of the line. The part number of each tubing assembly shall be identified. Color bands shall be plastic tape in accordance with Standard AND 10375."

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None

A DIVISION OF SENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO .: ZD- 22-322			DATE:			
CUSTOMER: Japan Air Lines		MCL 15.108 DTD 13 March 1961				
CHANGE NO. 72			MODEL: 22-22 (Convair	"880")		
TITLE: Cabin Rate o	f Climb Indicator,	Chan	nge to			
ORIGIN Japan Air Li 10 March 196	nes Letter SANQA-X	(-131,	, dated			
REASON FOR CHANGE:	Customer request					
EFFECT	ON WEIGHT *		EFFECT ON BALAN	ICE •		
GUAR, WT. EMPTY	OPER. WT. EMPTY					
Negligible	Negligible	Negligible		INCH LB.		
EFFECT ON GUARANTEED I	PERFORMANCE: * None					
ACCEPTANCE OF THIS CH	LL BE ACCUMULATED AND FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO THE FOLLOWING CHANGES	SAL N	Really 4/18/61	Misser		
			ENGINEERING	APPROVAL		
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		RECUI NON-R	CT ON PRICE PER AIRPLANE: RRING: ECURRING:			
ACCEPTED.		CONV	AIR, A DIVISION OF GENERAL DYNA	WICS CORP.		
BY:						
DATE				SUS ALIKAND AMERIKA		

Japan Air Lines Change No. 72

CONVAIR: SD

Title: Cabin Rate of Climb Indicator, Change to

Origin: Japan Air Lines Letter SANQA-X-131, dated 10 March 1961.

Reason for Change: Customer request

Description of Change:

Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below item under "PRESSURIZATION, HEATING AND AIR CONDITIONING" as follows:

Prom: "1 Indicator, Cabin Rate Kollsman D0-7165-10-055 of Climb

To: "1 Indicator, Cabin Rate Bendix-Pioneer 1653-6AB-A6-1"

To: "1 Indicator, Cabin Rate Bendix-Ploneer 1055-0AB-A0-1 of Climb

Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:				
CUSTOMER: Japan Air Lines			MCL 15,	107	Revis	ed rch 1961	
CHANGE NO: 71			MODEL: 22-22 (Convair "880"				
TITLE Crew Oxyge	en Mask, Change to						
ORIGIN Japan Air SANQA-X-17	Lines Letters: S. O dated 3-29-61.	ANQA-X	-116 da	ted 3-1	-61, and		
REASON FOR CHANGE:	Customer request						
	N WEIGHT *			EFFECT	ON BALANCE	*	
GUAR, WT. EMPTY	OPER, WT, EMPTY						
+1.0 1b	+1.0 lb			+	253	INCH LB.	
EFFECT ON GUARANTEED P	ERFORMANCE: *						
		None					
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	931	LUUL	HIYG I	PROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPLA	NES AFFE	CTED:			
SPECIAL PROVISIONS			T ON PRICE				
		NON-RE	CURRING: _				
ACCEPTED:					RAL DYNAMICS		
BY.							
DATE							

Japan Air Lines Change No. 71

CONVAIR: SD

Crew Oxygen Mask, Change to Title:

Japan Air Lines Letters: SANQA-X-116 dated 3-1-61, and SANQA-X-170 dated 3-29-61.

Reason for Change: Customer request

Description of Change:

Page A-16, APPENDIX I-C, OXYGEN EQUIPMENT:

Revise the below item under "Description" as follows:

From:	"5	Mask, Supplemental Oxygen	Sierra	358-158"
To:	"5	Mask, Supplemental Oxygen, Includes:	ARO	23808-100-DM
		(1) 5 Oxygen Hose (2) 5 Magnetic Mikes	ARO Carter	23808-120K CE-200D"

Effect on Weight Empty: +1.0 pound

Effect on Balance: +253 inch-pounds

Effect on Performance: None

The following shall not appear in the specification language:

Should the ARO 23808-120K oxygen hose not be suited for mating with the airplane connection, ARO will provide an oxygen hose which shall be compatible therewith.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:							
CUSTOMER: Japan Air Lines			MCL 15,031 DTD					
CHANGE NO: 70	- A	NODE	L: 22	-22	(Conv	air	"880")	
TITLE: Specification Administrative Transmitter P/N)			ve Change (Revision to Fuel Flow					
ORIGIN: Convair 1	nitiated							
REASON FOR CHANGE: S	ee statement at th	ne end	of t	this C	CP.		•	
EFFECT	ON WEIGHT *			Ei	FECT	Γ ON BAL	ANCE	*
GUAR, WT. EMPTY	OPER, WT. EMPTY							
0	0			,		0		INCH LB.
EFFECT ON GUARANTEED F								
	None	)				·		
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL	£	Att	EN EN	GINEERI	YG AP	PROVAL N.T.
LATEST DATE OF ACCEPTA	NCE:	AIRPLAI	NES A	FFECTE	D:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:							
ACCEPTED: BY:	CONVAIR	₹, A DI	IVISION O	F GEN	ERAL DY	NAMICS	CORP.	
DATE:				1 1	,		50474	(IR,5,0, 04117A

CONVAIR: SD

Japan Air Lines Change No. 70

Title: Specification Administrative Change (Revision to Fuel Flow

Transmitter P/N)

Origin: Convair initiated

Reason for Change: See statement at the end of this CCP.

Description of Change:

Page A-5, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below item under "ENGINE INSTRUMENTS - FUEL SYSTEM" as follows:

From: "4 Transmitter, Fuel Flow Gen. Elec. 8TJ59GAR-3"

To: "4 Transmitter, Fuel Flow Gen. Elec. 8TJ59GBB-3"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the specification language:

The language effect of CCP No. 32A (Installation of G.E.Fuel Flow Totalizer) as concerns the subject item is included in this paragraph.

The part number (8TJ59GAR-3) submitted for fuel flow transmitter in CCP No. 32A was for the basic transmitter, which when modified to make compatible with Japan Fuel Totalizer system is now redentified as G.E. 8TJ59GBB-3. This part number change is required to properly identify the units.



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIFFIG. CALIFORNIA

SPEC NO.: ZD- 22-322		DATE:				
CUSTOMER. Japan Air Lin	nes	MCL 15,030 DTD				
CHANGE NO: 69		MODEL: 22-22 (Convair "880")				
TITLE Specification A	Administrative	Chang	e (Landing Gear Pin	Indicator		
ORIGIN: Convair initiat	ced					
	specify that su lled.	bject	indicator lights a	re in-		
EFFECT ON WE	IGHT 4		EFFECT ON BALANCE *			
GUAR, WT. EMPTY OPE	R. WT. EMPTY					
0	0		O INC			
EFFECT ON GUARANTEED PERFO	ORMANCE: * Non	е				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Attendy	May partite		
			ENGINEER	ING APPROVAL		
LATEST DATE OF ACCEPTANCE:			AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE; RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED			AIR, A DIVISION OF GENERAL D	YNAMICS CORP.		
BY:						
DATE:				*05 (AT/), a. () Tal. (1)		

Title: Specification Administrative Change (Landing Gear Pin

Indicator Lights)

Origin: Convair initiated

Reason for Change: To specify that subject indicator lights are

installed.

Description of Change:

Page 22, Paragraph 3.8.1.5 LANDING GEAR SAFETY LOCK PINS:

Add the following to the end of the paragraph:

"Indicator warning lights shall be provided in the flight engineer's panel and shall be illuminated when the "safety lock" pins are inserted in the gear."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None



A CIVISION OF GENERAL DYNAMICS CORPORATION SAN DIL 40 CALEFORNIA

SPEC NO.: ZD- 22-322	DATE:			
CUSTOMER: Japan Air Lines	MCL 15,029 DTD			
CHANGE NO: 68	MODEL: 22-22 (Convair "880")			
Instrument Part Numbers)	ive Change (Revision to			
ORIGIN: Convair initiated				
REASON FOR CHANGE: See statement at 1	the end of this proposal			
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR, WT. EMPTY OPER, WT. EMPTY				
0 0	O INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *				
None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AT TOTALS REFLECTED IN A FUTURE CHANGE PROPACE OF THIS CHANGE IS DEPENDENT UPPRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	PON ES: HUC, MANNES:			
	ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:				
DATE:	TOWNAMA IN THE STREET			

Title: Specification Administrative Change (Revision to

Instrument Part Numbers)

Origin: Convair initiated

Reason for Change: See statement at the end of this proposal.

Description of Change:

Page A-4R, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below item under "SPERRY FLIGHT NAVIGATION SYSTEM" as follows:

From: "1 Flight Director Sperry 1780607

Computer (Z-5)

To: "1 Flight Director Sperry 1780607-1"

Computer (Z-5)

Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below item under "HYDRAULIC AND PNEUMATIC" as follows:

From: "1 Indicator, Emergency U.S. Gauge AW1826-AD02"
Air Brake Pressure

To: "1 Indicator, Emergency U.S. Gauge (P/N to be supplied Air Brake Pressure

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the specification language:

The change to part number of the flight director computer was established by the vendor as the current part number.

A change to part number of emergency air brake pressure indicato: is required due to change in pressure reading from 3,000 psi to 4,000 psi. This part number will be supplied when made available the vendor.

# CONVAIR A DIVINEN OF G NERAL DYNAMICS - ORPORATION SAN DIE GO, CALIFORNIA

SPEC NO. ZD- 22-322	DATE:			
CUSTOMER: Japan Air Lines	MCL 15,104 DTD			
CHANGE NO. 67A	MODEL: 22-22 (Convair "880")			
TITLE: Kollsman Integrated Fligh	it Instrument System, Change to			
ORIGIN: JAL Letter, SANQA-X-90, d	lated 20 February 1961			
REASON FOR CHANGE: Customer request	and revision to CCP No. 672.			
	*			
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR. WT. EMPTY OPER. WT. EMPTY				
0 0	O INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *  Non.	е			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROPERTY OF THIS CHANGE IS DEPENDENT UP PRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	OSAL ON 846 WASHIN			
	ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY.				
DATE	707164123754			

Title: Kollsman Integrated Flight Instrument System, Change to

Origin: JAL Letter, SANQA-X-90, dated 20 February 1961

Reason for Change: Customer request and revision to CCP No. 67.

Description of Change:

#### Page A-5, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "KOLLSMAN INTEGRATED FLIGHT INSTRU-MENT SYSTEM" as follows:

From:		Altimeter and Scale Error Corrector Set Control Chassis, in- cluding Computer	Kollsman Kollsman	A32667-10-001 A30720-00-001"
To:	*1	Altimeter and Scale Error Corrector Set Control Chassis, in- cluding Computer	Kollsman	A32667-10-001 B30720-00-004
	**1	Altimeter and Scale Error Corrector Set (drum-pointer-type) Control Chassis Assembly, Includes:	Kollsman Kollsman	A32647-10-001 B33280-00-004
	*	l Control Chassis 2 Computer Unit (for	Kollsman	B32140-00-001
	-	pilot and copilot) 8 Amplifier	Kollsman Kollsman	A30710-00-008 B27710-00-003"

Add the following note to the bottom of page:

"\*Effective Ships 1, 2 and 3.

\*\*Effective Ships 4 and 5."

The following shall not appear in the specification language:

Handbooks and manuals will not be revised to reflect the ship split effectivity for the above change.

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance: None



A II VISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPECINO: LU-EE-SEE			DATE:				
CUSTOMER: Japan Air	Lines		MCL 15,105 DTD 1 March 1961				
CHANGE NO: 66			MODEL: 22-22	(Convair	"880")		
TITLE: ATC Tran	sponder and DMET 1	System Control	, Change	to			
ORIGIN: JAL Letter, SANQA-X-101, dated 27 February 1961							
REASON FOR CHANGE:	Customer request						
	ON WEIGHT *		EFFECT ON BALANCE *				
GUAR, WT. EMPTY	OPER. WT. EMPTY						
0	0 .		0		INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: *  None							
NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH. PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	Attests	Duly.	One year		
			ENG	INEERING AF	PROVAL		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:			T ON PRICE PER AIRE RING: CURRING: :				
ACCEPTED		CONVA	IR, A DIVISION OF GENE	RAL DYNAMIC			
BY:							
DATE:	- 02-V	ALC INC. TO STATE OF					

CONVAIR: SD Japan Air Lines Change No. 66

Title: ATC Transponder and DMET System Power Control, Change to

Origin: JAL Letter, SANQA-X-101, dated 27 February 1961

Reason for Change: Customer request

Description of Change:

Page 75, Paragraph 3.17.1.7.1 POWER CONTROLS:

Revise the second and third sentences as follows:

From: "The "NORMAL" switch shall control the weather radar, the marker beacon, selective call systems No. 1 and No. 2, glide slope No. 2, navigation instrument transformer No. 2, VHF navigation No. 2, radio compass No. 2, VHF communication receiver No. 2, VHF communication transmitter No. 2, HF communication No. 2, DMET No. 1 and No. 2 if installed, and ATC transponder No. 1 and No. 2, if installed. The "ESSENTIAL" switch shall control the Edo Loran system, VHF communication receiver No. 1, VHF communication transmitter No. 1, VHF navigation No. 1, glide slope No. 1, radio compass No. 1, navigation instrument transformer No. 1, HF communication No. 1, and the Doppler system tracker and computer, if installed."

"The "NORMAL" switch shall control the weather radar, the marker beacon, selective call systems No. 1 and No. 2, glide slope No. 2, navigation instrument transformer No. 2, VHF navigation No. 2, radio compass No. 2, VHF communication receiver No. 2, VHF communication transmitter No. 2, HF communication No. 2, DMET No. 2 if installed, and ATC transponder No. 2, if installed. The "ESSENTIAL" switch shall control the Edo Loran system, VHF communication receiver No. 1, VHF communication transmitter No. 1, VHF navigation No. 1, glide slope No. 1, radio compass No. 1, navigation instrument transformer No. 1, HF communication No. 1, DMET No. 1 if installed, ATC transponder No. 1 if installed, and the Doppler sensor and the Doppler navigation computer, if installed."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the specification language:

The language effects of CCP No. 54 (Revision to Par. 3.17.1.7.1) and CCP No. 61 (Various Changes to Electronic Items Language) are included in this proposal. CCP No. 54 has been approved by Customer; however, this proposal is based on prior Customer approval of CCP No. 61.

A DI ZISION OF GENERAL DYNAMICS CORPORATION SAN DIL GO, CALIFORNIA

SPEC NO.: ZD-22-322		DATE:				
CUSTOMER Japan Ai	r Lines	MCL 15,016 DTD 1 March 1961				
CHANGE NO: 65		MODEL: 22-22 (Convair "880")				
TITLE: Loran Rece	iver Connection, I	leloca	tion of			
ORIGIN. JAL Letter	, SANQA-X-102, dat	ed 27	February 1961			
REASON FOR CHANGE:	Customer request					
EFFECT	ON WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER, WT. EMPTY					
0	0		O INCH LB.			
EFFECT ON GUARANTEED	PERFORMANCE: *					
	None					
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSA ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			politically 4/4/. With			
			ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS			EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED:			IR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY!						
DATE						

CONVAIR: SD

Title: Loran Receiver Connection, Relocation of

Origin: JAL Letter SANQA-X-102, dated 27 February 1961.

Reason for Change: Customer request

Description of Change:

The specification language is not effected by this change which proposes to relocate the Loran Receiver antennaled from No. 1 HF Coupler Accessory Unit to No. 2 HF Coupler Accessory Unit.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

jih day o

# CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:			
CUSTOMER: Apan Air Lines			MCL 15,028 DTD			
CHANGE NO: 64			MODEL: 22-22	(Convair	"880")	
TITLE: Specificat and Revisi	ion Administrative on to Oxygen Regul	e Chan lator	ge (Typographi P/N)	cal Corre	ctions	
ORIGIN: Convair in	itiated					
	To correct typograpert number for ox Vendor.	aphica cygen	l errors and t regulator as e	o provide stablished	current i by	
	ON WEIGHT *		EFFEC	T ON BALANCI	E *	
GUAR, WT. EMPTY	OPER. WT. EMPTY					
0	0			0	INCH LB.	
EFFECT ON GUARANTEED F	PERFORMANCE: * None					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			1	Lapy Mark	-21	
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED:		CONVA	IR, A DIVISION OF GENI	ERAL DYNAMIC	S CORP.	
DATE.				tou v	413,5.D. 6-1217A	

# CONTAIN

22-322	D/T.
Jopan Air Lines	4   15,102   15 February
	MQUEL 22-22 (Convair "080 )
VHF Reselvers, Change to	
Japan Air Lines verbal reques	st on 16 Februar, 1961
As above	
-THE TON WEIGHT	EFFECT ON EALAND
INPER WT EMPTY	The state of the s
0	0
None	
THIS CHANGE IS THE PROPOSED OF THE POLITION OF	IAL DOLL
ALCEP FARCE	AIRPLANES AFFECTEE
	PETICION PARTICER ARULTO RECURRING NON PETIREINT TOTAL

VHF Receivers, Change to

Japan Alr Lines vertal request on 16 February 1961

Theson for Change: As above

Telegraption of Change:

#### A-9, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "VHF NAVIGATION SYSTEM" as Tollows:

Prom: "2 Receiver (AC Power) Collins 51X-2"

To: "2 Receiver (522-0833-064, AC Power) Collins 51X-2"

Revise the below item under "VHF COMMUNICATION" as follows:

From: "2 Receiver (AC Power) Collins 51X-2"

To: "2 Receiver (522-0833-064, AC Power) Collins 51X-2"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:				
CUSTOMER: Japan Air Lines			MCL 15,024 DTD				
CHANGE NO: 58			MODEL: 22-22 (Convair "880")				
TITLE: Specific Seat Back	cation Administrat ck Movement)	ive C	Change (Revision to Passenger				
ORIGIN: Convair	initiated						
REASON FOR CHANGE:	To make specifica seat back position	tion	language agree with actual capability.				
EFFECT (	ON WEIGHT *		EFFECT ON BALANCE *				
GUAR, WT. EMPTY	OPER. WT. EMPTY						
0	0		O INCH LB.				
EFFECT ON GUARANTEED P	EFFECT ON GUARANTEED PERFORMANCE: *  None						
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSA ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Attenting 720/6, WY 212 MM ENGINEERING APPROVAL				
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS.			EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED			AIR, A DIVISION OF GENERAL DYNAMICS CORP.				
BY:							
DATE:			TON ANY S D. GREET A				

CONVAIR: SD Japan Air Lines Change No. 58

Title: Specification Administrative Change (Revision to Passenger

Seat Back Movement)

Origin: Convair initiated

Reason for Change: To make specification language agree with actual

seat back positioning capability.

Description of Change:

Page 83, Paragraph 3.19.1.1.6.1 SEAT BACK MOVEMENT:

Starting with the third sentence ("The seat back shall fold further .....") delete the remainder of the paragraph and substitute the following:

"The seat back shall fold to approximately 30 degrees forward of the vertical position by applying a load of not less than 30 pounds, or more than 35 pounds, at the top of aft side of the seat back without removing the center arm rest. The seat back may be further folded to approximately a horizontal position by removing the center arm rest. Seat back positions shall be as follows:

#### First Class Seats:

Normal Upright: 15 degrees aft of vertical

Adjustable Stops at: 38 and 45 degrees aft of vertical

Coach Seats:

Normal Upright: 15 degrees aft of vertical

Adjustable Stops at: 23 and 38 degrees aft of vertical"

Effect on Weight Empty: O
Effect on Balance: O
Effect on Performance: None

A 1 IS OR OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:		
CUSTOMER: Japan A:	ir Lines		MCL 15,101	DTD 24	January 1
CHANGE NO: 57			MODEL: 22-22	(Convair	"380")
TITLE: Seat Arm Rest	Section Aft of E	merger	ncy Escape Ha	tch; Remo	val of
ORIGIN. Japan Air Li	nes request	and the second			
REASON FOR CHANGE:	s above				
REFECT O	N WEIGHT *	101 (-01)	EFFE	CT ON BALAN	ICE *
GUAR, WT. EMPTY	OPER. WT. EMPTY				
-5.0 lbs	-5.0 lbs.		-3,53	5	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *				
	None				
NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TI CCP No. 49	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	Atten	the the	/ dis   61
Т но. ту	3		E	NGINEERING	APPROVAL
LATEST DATE OF ACCEPTAI	VCE:	AIRPL	ANES AFFECTED:		
REC NOI		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL.			
ACCEPTED:		CONVA	IR, A DIVISION OF GE	NERAL DYNAM	ICS CORP.
LY	***************************************				
FIA I F.					

Title: Seat Arm Rest Sections Aft of Emergency Escape Hatch, Removal of

Origin: Japan Air Lines requested

Reason for Change: As above

Description of Change:

Page 104, Paragraph 3.23.2 - EQUIPMENT INTERCHANGEABILITY:

Delete the asterisks from the following items under "a.

Interchangeable Parts"

"\*\*Passenger Seat Assembly R-H
\*\*Passenger Seat Assembly L-H"

Delete the following note from the bottom of Page 104:

"\*\*NOTE: Passenger seats immediately aft of the escape hatches shall have special outboard arm rests; however they will be interchangeable in all other respects".

Effect on Weight Empty: -5.0 lbs.

Effect on Balance: -3,535 in. lbs.

Effect on Performance: None

The following shall not appear in the specification language.

"This proposal deletes the integral arm rest sections on the fuselage in the area aft of the emergency escape hatches. The two outboard seats (one L.H. and one R.H.) just aft of the emergency escape hatches, will be replaced with typical coach seats with standard arm rests; thereby making them interchangeable with existing similar seats. This proposal is based on coach seats being presently installed as defined in CCP No. 49 (Modification of Passenger Seating Configuration); therefore, this change is contingent on prior acceptance of CCP No. 49".



A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: 20- 22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,023 DTD			
CHANGE NO: 56		MODEL: 22-22 (C	onvair '	'880")		
	ion Administrative and 3.11.8)	c Char	nge (Revision to	Paragra	iphs	
ORIGIN: Convair in	itiated					
REASON FOR CHANGE: T	o clarify the inte	ent of	subject paragr	aphs.		
EFFECT	ON WEIGHT *		EFFECT (	ON BALANC	E *	
GUAR, WT, EMPTY	OPER. WT. EMPTY					
0	0			0	INCH LB.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None					
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Allerte	AGICY INEERING A		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
RE		EFFECT ON PRICE PER AIRPLANE; RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED:		CONVA	AIR, A DIVISION OF GENER	RAL DYNAMIC	CS CORP.	
BY:						
DATE:				7.0%	5-4100, at 0-12174	

CONVAIR: SD

Title: Specification Administrative Change (Revision to Paragraphs

3.10.1.5.1 and 3.11.8)

Origin: Convair initiated

Reason for Change: To clarify the intent of subject paragraphs.

Description of Change:

Delete the last two sentences and substitute the following:

"A switch, guarded in the on position, shall be installed on the pedestal to shut off all hydraulic power to the stabilizer screw jack. A three-position momentary switch shall be installed on the pedestal which, after the hydraulic shutoff switch is moved to "OFF", operates the standby electrical-mechanical stabilizer trim control. In addition, hand cranks shall be installed on the pedestal for emergency manual stabilizer trim control."

Page 30, Paragraph 3.11.8 INSPECTION AND MAINTENANCE:

Delete the last sentence and substitute the following:

"Each engine QEC (quick engine change build-up) shall be interchangeable and replaceable as a complete assembly, except for reverser cascades, and certain accessories which may not be required on individual installations."

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

CONVAIR: SD

1 m 3

Title: Specification Administrative Change (Typographical Cor-

rections and Revision to Oxygen Regulator P/N)

Origin: Convair initiated

Reason for Change: To correct typographical errors and to provide

current part number for oxygen regulator as

established by Vendor.

Description of Change:

Page 2 Paragraph 2.3 APPLICABLE CONVAIR SPECIFICATION:

In second line from bottom of page, change "Electroni" to "Electronic".

Page 11, Paragraph 3.2.6 OPERATING TEMPERATURE:

In second line of first sentence, change "soacked" to "soaked".

Page 46, Paragraph 3.14.3.2.1 STATIC SOURCES:

In first line of first sentence, change "loacted" to "located".

Page A-16, APPENDIX I-C, OXYGEN EQUIPMENT:

Revise the below item as follows:

From: "5 Regulator ARO Equipment 15830"

To: "5 Regulator ARO Equipment 15830A"

Effect on Weight Empty: 0

Elfa-

Effect on Balance: 0
Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE 50, CALIFORNIA

CUSTOMER. Japan Air Li							
CUSTOMER: Japan Air Lines			MCL 15,103	MCL 15,103 DTD 24 February 190			
CHANGE NO: 63		MODEL: 22-22 (Convair "880")					
TITLE: Tie-Down Bolts	, Reduction of	in No	o. 2 Buffet				
ORIGIN Japan Air Line dated 23 Februa		A-X-98	3), Mitsuoka	to Chamber	es,		
REASON FOR CHANGE: Cust	omer request						
EFFECT ON WE	IGHT *		EFFE	CT ON BALANC	E *		
GUAR. WT. EMPTY OPE	R. WT. EMPTY						
Negligible	Negligible		Ne	gligible	INCH LB.		
EFFECT ON GUARANTEED PERFO	DRMANCE: *	ne		•			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Jul	who	Yn/c.		
				NGINEERING A	APPROVAL Ys		
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED:		CONVA	IR, A DIVISION OF GE	ENERAL DYNAM	CS CORP.		
BY.							
DI.							

Tie-Down Bolts, Reduction of in No. 2 Buffet Title:

Japan Air Lines Letter (SANQA-X-98), Mitsuoka to Chambers, Origin:

dated 23 February 1961

Reason for Change: Customer request

Description of Change:

This proposal has no effect on specification language which reduces the number of tie-down bolts in No. 2 Buffet from 24 to 18.

Effect on Weight Empty: Negligible Effect on Balance: Effect on Performance: Negligible

None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322		DATE:					
CUSTOMER: Japan Air Lines			MCL 15,027	DTD			
CHANGE NO: 62			MODEL: 22-22 (Convair "880")				
TITLE. Specificatio and to Appen	n Administrative (dix I-C)	hange	(Revision to	Par. 3.19.	1.1.3		
ORIGIN: Convair init	iated To clarify the int	tent o	of the specific	cation.			
EFFECT (	ON WEIGHT *		EFFEC	T ON BALANCE	*		
GUAR, WT. EMPTY	OPER. WT. EMPTY				<u> </u>		
0	0			0	INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: *						
	None						
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF TI	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N		JAM,	Jung Zielu PROVAL		
LATEST DATE OF ACCEPTAN	NCE:	AIRPL	ANES AFFECTED:				
SPECIAL PROVISIONS:		RECUR NON-RE	T ON PRICE PER AIR RING: CURRING:				
ACCEPTED:  BY:  DATE:	-	CONVA	IR, A DIVISION OF GEN				
				10.1 A VI	R. 7. D. 6-121 - A		

CONVAIR: SD

Title: Specification Administrative Change (Revision to Par. 3.19.1.1.3

and to Appendix I-C)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification

Description of Change:

Page 83, Paragraph 3.19.1.1.3 - OBSERVER'S SEAT:

Revise the first sentence as follows:

From: "A folding-type seat aft of the third pilot shall be

installed for use by an observer".

"A folding-type seat shall be installed aft of the navi-To:

gator's seat for use by an observer".

Page A-4R, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Delete the below items under "FLIGHT INSTRUMENTS:"

Indicator Light Power Failure

Detector Unit, Power Failure 2

Convair

Convair

Page A-8, RIECTRICAL EQUIPMENT:

Add the following item under "ELECTRICAL POWER EQUIPMENT":

"2 Indicator Light, Power Failure

Korry

ST753-E6

Effect on Weight Empty: Rffect on Balance:

Effect on Performance: None

The following shall not appear in the specification language:

The revision to Paragraph 3.19.1.1.3 corrects an error which was inadvertently incorporated during preparation of the specification.

The relocation of the power failure indicator lights from "FLIGHT INSTRUMENTS" to "ELECTRICAL POWER EQUIPMENT", places this item in its proper category.

The power failure detector unit is a non-existent item which is proposed for deletion for purpose of specification clarification.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:			
CUSTOMER Japan A	ir Lines	MCL 15,026 DTD				
CHANGE NO: 61			MODEL: 22-22 (Convair "880")			
TITLE Specif Electr	ication Administra onic Items Languag	tive e)	Change (Various Changes to			
	r initiated based 20 and 27 February		pan Air Lines letters			
REASON FOR CHANGE:	As above					
EFFECT	ON WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER. WT. EMPTY					
0	0		O INCH LB.			
EFFECT ON GUARANTEED	PERFORMANCE; *	3				
	None					
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Attendy of 3/14/6, White approval NO			
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS.		EFFECT ON PRICE PER AIRPLANE:				
		NON-R	ECURRING:			
ACCEPTED		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY						
DATE:			_ 250 9 6 ( 5 , 5 , 6 , 1 , 7 )   4			

CONVAIR: SD Japan Air Lines
Change No. 61

Page 1 of 4

Title: Specification Administrative Change (Various Changes to Elec-

tronic Items Language)

Origin: Convair initiated, based on Japan Air Lines Letters dated 20

and 27 February 1961.

Reason for Change: As above

Description of Change:

Page 46, Paragraph 3.14.2.2 INTERCONNECTION OF EQUIPMENT:

Delete the sentence under above paragraph title, and add the following new paragraphs:

"3.14.2.2.1 COMPASS INTERCONNECTION: Pilot's RMDI (C-6B) shall function as part of the compass -1 system, and shall be controlled by a servo amplifier included in the horizon -1 system integrated instrument rack, and supply heading signals to the following systems and instruments:

Copilot's RMI (radio magnetic indicator)
Navigator's RMI
VOR-1 instrumentation unit
Pilot's PDI (Pictorial deviation indicator)
Pilot's PDI course selector
Navigation radar
Flight data recorder

The heading selector of pilot's RMDI (C-6B) shall supply selected heading signal to the automatic pilot. The directional gyro of compass -1 system shall supply heading reference signal to the automatic pilot. Copilot's RMDI (C-6B) shall function as part of the compass -2 system, and shall be controlled by a servo amplifier included in the horizon -2 system integrated instrument rack, and shall supply heading signal to the following systems and instruments:

Pilot's RMI Navigator's RMI VOR-2 instrumentation unit Copilot's PDI

The heading selector of copilot's RMDI (C-6B) shall supply selected heading signal to the flight director system. The compass transfer switch shall be installed in navigator's panel to make selection of the heading signal to navigator's RMI from either compass system.

- 3.14.2.2.2 HORIZON INTERCONNECTION: Pilot's HZ-4 horizon flight director indicator shall function as part of the horizon -1 system, and shall be controlled by two servo amplifiers in the horizon -1 system integrated instrument rack. The verical gyro of the horizon -1 system shall supply signal to the automatic pilot. Copilot's HZ-4 horizon flight director indicator shall function as part of the horizon -2 system, and shall be controlled by two servo amplifiers in the horizon -2 system integrated instrument rack. The vertical gyro of the horizon -2 system shall supply signal to the flight director system, and the weather radar system. The gyro trim knob of the pilot's and copilot's horizon flight director indicator shall position the attitude sphere of pilot's and copilot's horizon flight director indicator respectively.
- 3.14.2.2.3 FLIGHT DIRECTOR INTERCONNECTION: The remote heading selector knob of the flight reference selector shall position the heading selector of both RMDI (C-6B) with two servo amplifiers, one in each of the compass racks. Automatic crab angle, following in the approach mode, shall be provided by one servo amplifier in compass -1 system compass rack. One spare servo amplifier shall be included in compass -2 system compass rack. (See 3.17.3.4.4.)
- 3.14.2.2.4 PDI (R-1) INTERCONNECTION: Pilot's PDI (R-1) shall be controlled by one servo amplifier and one "to-from" sensor in the horizon -1 system integrated instrument rack. Copilot's PDI (R-1) shall be controlled by one servo amplifier and one "to-from" sensor in the horizon -2 system integrated instrument rack. (See 3.17.3.4.3.)
- Page 66, Paragraph 3.16.10.8 MAINTENANCE RECEPTACLES:

  Add the following sentence to the end of the paragraph:

  "The type of receptacle shall be the same as Hubbell 7528."
- Page 75, Paragraph 3.17.1.7.1 POWER CONTROLS:

In the third sentence, change the last nine words as follows:

From: "and the Doppler system tracker and computer, if installed."

To: "and the Doppler sensor and the Doppler navigation computer, if installed."

Page 3 of 4

#### Page 78, Paragraph 3.17.2.4 AIRBORNE SELECTIVE CALLING:

Add the following to the end of the paragraph:

"The lights on the control panel shall be connected to the steady output of the Selcal systems. Aural calling shall be provided by means of a chime connected to the interrupted output of the Selcal systems. The pilot's call bell shall not be used for the Selcal systems."

#### Page 78, Paragraph 3.17.3.1.2 CONTROL:

Delete the last sentence "(The heading data shall be connected to the two RMDI (C-6B) indicators and the one RMI indicator at the navigator's station)" and substitute the following:

"The single and double pointers on pilot's and copilot's RMDI (C-6B) and navigator's RMI shall display information from the No. 1 and No. 2 ADF receivers respectively."

#### Page 79, Paragraph 3.17.3.4.1 CONTROLS:

Delete all description after the second sentence, and add the following new paragraphs, after Paragraph 3.17.3.4.2:

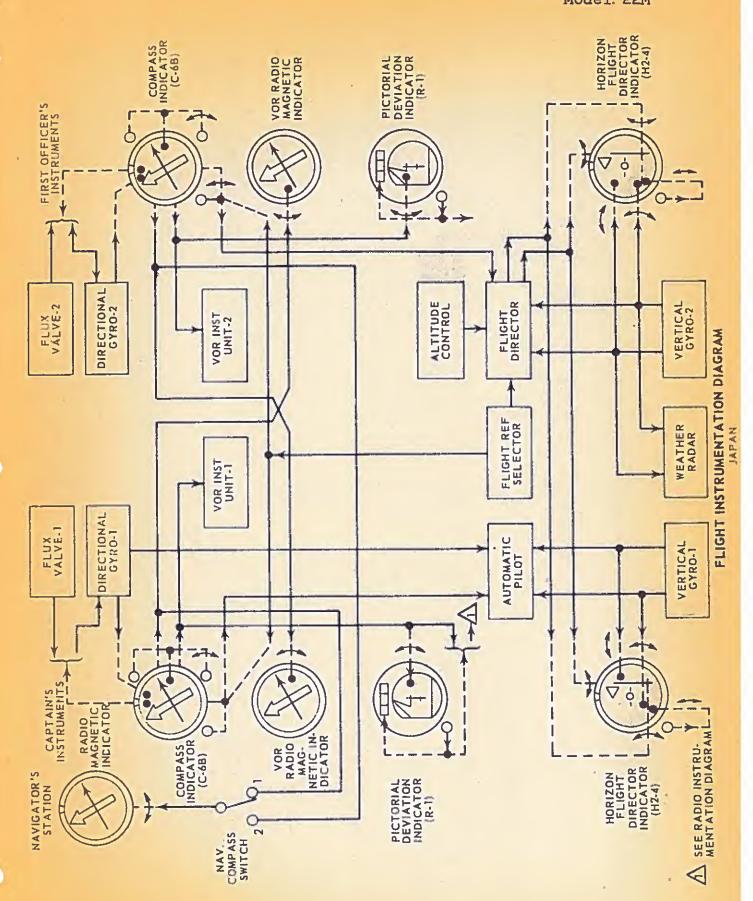
- "3.17.3.4.3 INSTRUMENTATION: Pilot's PDI (R-1) shall display information from No. 1 VHF navigation system and No. 1 glide slope, whereas copilot's PDI shall display information from the No. 2 VHF navigation system and No. 2 glide slope. The course selector of pilot"s PDI shall supply signal to the No. 1 VHF navigation system, whereas the course selector of copilot's PDI shall supply signal to the No. 2 VHF navigation system The single and double pointers on pilot's and copilot's RMI shall display information from the No. 1 and No. 2 VHF navigation systems respective. The automatic pilot shall receive deviation and mode control signals from the No. 1 VHF navigation system and No.1 glide slope, and shall receive VOR slected course signals from pilot's PDI.
  - 3.17.3.4.4 VHF NAVIGATION SIGNAL FOR FLIGHT DIRECTOR: A flight director transfer switch with "VOR/ILS-1" and "VOR/ILS-2" positions shall be installed. Two flags, each with "VOR/ILS-1" and "VOR/ILS-2" positions shall be installed. (See Page 44a.) When the transfer switch

Page 4 of 4

is in the "VOR/ILS-1" position, the flight director system shall be connected to the No. 1 VHF navigation system and No. 1 glide slope, and both flags shall indicator "VOR/ILS-1" position. When the transfer switch is in the "VOR/ILS-2" position, the flight director system shall be connected to the No. 2 VHF navigation system and No. 2 glide slope, and both flags shall indicate "VOR/ILS-2" positions."

- Enclosures: (A) One copy Page 46b FLIGHT INSTRUMENTATION DIAGRAM (To be incorporated into specification.)
  - (B) One copy Page 46c RADIO INSTRUMENTATION DIAGRAM
    (To be incorporated into specification.)

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None



STATION

- V ISUM OF ALNERAL DYNAMICS COMPONATION HAM UIE 30, CALIFORNIA

SPEC NO. ZD- 22-322			DATE:		
CUSTOMER Japan A1	r Lines		MCL 15,095	DTD 8 No	ised vember 196
CHANGE NO: 55			MODEL: 22-22	(Convair	"880")
TITLE Customer	Furnished Equipme	nt, I	nstallation o	of	
ORIGIN JAL Memo	No. 13 (Mitsuoka Customer request	to Du	ckworth) date	ed 31 Octo	ber 1960.
	ON WEIGHT *		EFFE	CT ON BALANO	_t '
OPER. WT. EMPTY O -14.0			-6,	,075	INCH LB.
EFFECT ON GUARANTEED P		ne			
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE	AL V	Atten	THE MISSING A	APPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED			IR, A DIVISION OF GE		
вү.					
FIATE					

Page 1 of 2

Title: Customer Furnished Equipment, Installation of

Origin: JAL Memo No. 13 (Mitsuoka to Duckworth) dated 31 October 1960.

Reason for Change: Customer request

Description of Change:

Page 80, Paragraph 3.17.3.6 EMERGENCY RADIO TRANSMITTER:

Revise the sentence to read as follows:

"Stowage shall be provided in the forward right hand coat compartment for one Customer furnished AN/CRT-3 emergency radio transmitter."

Page 88, Paragraph 3.19.2.6.2 SIGNAL PISTOL:

From: "Provisions for the stowage of one Kilgore G-37-K signal pistol and six Kilgore 3-37 shells shall be made in the pilots' compartment."

To: "Stowage shall be provided in the pilots' compartment for one Kilgore G-37-K signal pistol and six Kilgore 3-37 shells.

Page 94, Paragraph 3.19.7.1 LIFE RAFTS:

Revise the first sentence to read as follows:

"The installation of five Customer furnished, 25-man life rafts shall be made in the following locations."

Page 94, Paragraph 3.19.7.2 EMERGENCY TRANSMITTER:

Revise the sentence to read as follows:

"Stowage shall be provided for an emergency transmitter as specified in 3.17.3.6."

Page A-1, APPENDIX I-A, CUSTOMER FURNISHED - CONVAIR INSTALLED:

Add the following items under "Description":

"* .97 * 5	Life Vests			170.0 755.0
	Life Rafts Emergency Radio		AN/CRT-3	40.0
* 1	Transmitter Signal Pistol	Kilgore	G-37-K	\7.0"
* 6	Shell, Signal Pistol	Kilgore	3-37	(1.0

Page 2 of 2

Add the following to the bottom of Page A-1:

"\*Useful Load Items"

#### Page A-2, APPENDIX I-B, CUSTOMER FURNISHED - CUSTOMER INSTALLED:

### Delete the following items under "Description":

11#1	05	Life Ve	ests				184.0
*	5	Life Ra	afts				755.0
*	1	Emerger	ncy Rad:	io			
		Trans	smitter			AN/CRT-3	40.0
#	1	Signal	Pistol		Kilgore	G-37-K	\$7.0
*	6	Shell,	Signal	Pistol	Kilgore	3-37	(1.0

Delete the following from the bottom of Page A-2:

Effect on Weight Empty:

Effect on Operating Weight Empty: -14.0 pounds

Effect on Balance: Effect on Performance: -6,075 inch-pounds

None

<sup>&</sup>quot;\*Useful Load Items"

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

PEC NO.: ZD-22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,022 DTD			
CHANGE NO: 54	CHANGE NO: 54			vair "880")		
TITLE: Specificat 3.17.1.7.1		e Chai	nge (Revision to	Paragraph		
ORIGIN: Convair in		ntent	of Paragraph 3.1	.7.1.7.1.		
EFFECT	ON WEIGHT *		EFFECT ON	BALANCE *		
GUAR. WT. EMPTY	OPER. WT. EMPTY					
0	0		0	INCH LB.		
EFFECT ON GUARANTEED F	PERFORMANCE: * None					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Hady	13161 log 13/4		
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED:		CONVA	IR, A DIVISION OF GENERAL	L DYNAMICS CORP.		
BY:						
DATE				-9- V 50's		

CONVAIR: SD

Title: Specification Administrative Change (Revision to Paragraph

3.17.1.7.1)

Origin: Convair initiated

Reason for Change: To clarify the intent of Paragraph 3.17.1.7.1.

Description of Change:

Page 75, Paragraph 3.17.1.7.1 POWER CONTROLS:

Delete the paragraph and substitute the following:

POWER CONTROLS: One "NORMAL" and one "ESSENTIAL" "3.17.1.7.1 switch, located accessible to both pilots, shall control the ac and dc power to the radio equipment. The "NORMAL" switch shall control the weather radar, the marker beacon, selective call systems No. 1 and No. 2, glide slope No. 2, navigation instrument transformer No. 2, VHF navigation No. 2, radio compass No. 2, VHF communication receiver No. 2, VHF communication transmitter No. 2, HF communication No. 2, DMET No. 1 and No. 2 if installed, and ATC transponder No. 1 and No. 2, if installed. The "ESSENTIAL" switch shall control the Edo Loran system, VHF communication receiver No. 1, VHF communication transmitter No. 1, VHF navigation No. 1, glide slope No. 1, Radio compass No. 1, navigation instrument transformer No. 1, HF communication No. 1, and the Doppler system tracker and computer, if installed. The emergency dc bus, through suitable circuit breakers, shall power the audio selector panels, the public address amplifier, the interphone, the pilot compartment speaker, the flight data recorder, if installed, and the tape reproducer, if installed. All No. 1 normal radio equipment shall be supplied by a separate ac and dc line. All No. 2 normal radio equipment shall be supplied by a second separate ac and dc line. Protection shall be provided for each line. The power supply, feeder wires, circuit protection for equipment which back up each other functionally shall be separated in a manner that will preclude operation failure of both if any one circuit protective device feeder wire or power supply fails. Illumination shall be provided for the circuit breakers.

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,021 DTD			
CHANGE NO: 53A			MODEL: 22-22 (Convair "880")			
	ation Administrati tems Language)	lve Ch	ange (Misc. 1	Revisions	to Elec-	
ORIGIN: Japan A1	r Lines requested					
REASON FOR CHANGE:	Revision to CCP N	No. 53	•			
EFFECT	ON WEIGHT *		EFFE	CT ON BALANC	CE *	
GUAR, WT. EMPTY	OPER, WT. EMPTY					
0	0			0	INCH LB.	
EFFECT ON GUARANTEED F	ERFORMANCE: * None					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOS ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL	pote	engineering.	APPROVAL	
LATEST DATE OF ACCEPTA	NCE.	AIRPI	ANES AFFECTED:			
LATEST DATE OF ACCEPTANCE:			ANLO AL LECTLE.			
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.				
BY:						
DATE:					= \$1 t, > 6-1. 1	

CONVAIR: SD

Japan Air Lines Change No. 53A

Page 1 of 2

Title: Specification Administrative Change (Misc. Revisions to Elec-

tronic Items Language)

Origin: Japan Air Lines requested

Reason for Change: Revision to CCP No. 53.

Description of Change:

Page 71, Paragraph 3.17.1.1 CONTROL PANELS:

At top of page, add the following after "Weather Radar":

"(1)"

Change the below items under "Provisions for the following control panels" as follows:

From: "Doppler Computer Doppler Navigation"

To: "Doppler Sensor
Doppler Navigation Computer"

Page 78, Paragraph 3.17.3.1.2 CONTROL:

Revise the second sentence as follows:

From: "The heading data shall be connected to the RMI indicators."

To: "The heading data shall be connected to the two RMDI (C-6B) indicators and the one RMI indicator at the navigator's station."

Page 79, Paragraph 3.17.3.4.1 CONTROLS:

Revise the fourth and fifth sentences as follows:

"The single and double pointers on the RMDI shall display information from the No. 1 and No. 2 Navigation Receivers respectively. The single and double pointers on the RMI shall display information from the No. 1 and No. 2 ADF receivers respectively."

To:

"The single and double pointers on the RMI (VOR) on the pilot's and copilot's panels shall display information from the No. 1 and No. 2 Navigation Receivers respectively. The single and double pointers on the RMDI (C6B) on the

CONVAIR: SD

Japan Air Lines Change No. 53A

Page 2 of 2

pilot's and copilot's panels and RMI at navigator's station, shall display information from the No. 1 and No. 2 ADF receivers respectively."

# Page 80a, ANTENNA ARRANGEMENT:

Above illustration will be revised to delete the word "provisions" from "HF ANTENNA COUPLER NO. 2" callout.

Effect on Weight Empty: O
Effect on Balance: O
Effect on Performance: None

CONVAIR: SD

Specification Administrative Change (Revision to Auxiliary Landing Gear, Tail Skid) Title:

Convair initiated Origin:

Reason for Change: Product improvement.

Description of Change:

Page 24, Paragraph 3.8.3 AUXILIARY LANDING GEAR (TAIL SKID):

Change the first line in the paragraph to read as follows:

"A retractable skid or ......" From:

To: "A skid or ....."

Effect on Weight Empty: 0 Effect on Balance: 0 Effect on Performance:

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,100	DTD 14	December 1	
CHANGE NO: 51			MODEL: 22-22	(Convair	"880")	
TITLE: JAL Buf:	fet Inserts, Chan	ge fro	om Customer to	Convair	Installed	
ORIGIN: JAL Memo	o No. 23 (Mitsuoka	a to I	Duckworth) date	ed 22 Nov	ember 1960	
REASON FOR CHANGE:	Customer request					
EFFECT (	ON WEIGHT *		EFFEC	T ON BALAN	ICE *	
GUAR. WT. EMPTY	OPER, WT. EMPTY					
0	0		0		INCH LB.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL  ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:  ENGINEERING APPROVAL						
LATEST DATE OF ACCEPTANCE:			AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED:		CONVA	IR, A DIVISION OF GEN	IERAL DYNAM	ICS CORP.	
BY:						
DATE:	•			-/>	WATE	

Title: JAL Buffet Inserts, Change from Customer to Convair Installed

Origin: JAL Memo No. 23 (Mitsuoka to Duckworth) dated 22 November 1960

Reason for Change: Customer request

Description of Change:

Page A-1, APPENDIX I-A, CUSTOMER FURNISHED - CONVAIR INSTALLED:

Add the following item under "Description"

"\*AR Inserts, Buffet"

Add the following to the bottom of Page A-1:

"\*Useful Load Item"

Page A-2, APPENDIX I-B, CUSTOMER FURNISHED - CUSTOMER INSTALLED:

Delete the below item under "Description"

"\*AR Inserts, Buffet"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the specification language.

It is to be understood that Convair in no way will be responsible for inspection, quality, warranty or certification and loss, design deficiencies or fitting of the Customer furnished buffet inserts connected with this proposal.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322		DATE:				
CUSTOMER: Japan Air Lines			MCL 15,097 DTD 5 December 1960			
CHANGE NO: 50			MODEL: 22-22 (Convair "880")			
TITLE: Portable Oxygen Bottles, Increase from 6 to 7						
ORIGIN: JAL Memo No. 26 (Mitsuoka to Duckworth) dated 5 December 1960  REASON FOR CHANGE: Customer request						
EFFECT ON WEIGHT *			EFFECT ON BALA	NCE *		
GUAR. WT. EMPTY Negligible	OPER, WT. EMPTY +7.0 lbs.		+2,954	INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: *  None						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL  ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:  ENGINEERING APPROVAL						
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED: C			AIR, A DIVISION OF GENERAL DYNA	AMICS CORP.		
BY:						
DATE:	4			ON VALIDO 0-1217 M		

CONVAIR: SD

Title: Portable Oxygen Bottles, Increase from 6 to 7.

Origin: JAL Memo No. 26 (Mitsuoka to Duckworth) dated 5 December 1960.

Reason for Change: Customer request

Description of Change:

Page 94, Paragraph 3.19.5.6 - PORTABLE BOTTLES:

Revise the second sentence as follows:

"Six, seven-cubic foot portable oxygen bottles shall be provided, three located in the aft L.H. coat closet, two in the forward R.H. coat closet and one in the forward L.H. coat closet".

"Seven, seven-cubic foot portable oxygen bottles shall be provided; three located in the aft L.H. coat closet, three in the forward R.H. coat closet and one in the forward L.H. coat closet".

Page A-15, APPENDIX I-C, OXYGEN SYSTEM:

Revise the below items as follows:

From:	°6	*Bottle, Portable Oxygen (7 cu ft)	Scott	5500-B1A-BF20B 86.0"
	6	Mask, Oxygen (ASB)	Scott	23275
To:	"7	*Bottle, Portable Oxygen (7 cu ft)	Scott	5500-B1A-BF20B 93.0
	7	Mask, Oxygen (A8a)	Scott	23275

Effect on Weight Empty: Negligible

Effect on Operating Weight Empty: +7.0 lbs.

Effect on Balance: +2,954 in. lbs.

Effect on Performance: None

A D VILLEN OF GENERAL DYNAMICS CORPORATION SAN DIE ED, CALLFORINA

SPEC NO. ZD- 22-322	DATE:				
CUSTOMER Japan Air Lines	MCL 15,096 DTD 5 December 196				
CHANGE NO: 49	MODEL: 22-22 (Convair "880")				
TITLE Passenger Seating Configur	ration, Modification of				
ORIGIN: JAL Memo No. 26 (MITSUOKA to DUCKWORTH) dated 5 December 1960					
REASON FOR CHANGE Customer request					
EFFECT ON WEIGHT *	EFFECT ON BALANCE				
GUAR, WT. EMPTY OPER, WT. EMPTY					
+40.0 lb +26.0 lb	-34,151 INCH LB.				
EFFECT ON GUARANTEED PERFORMANCE: 1	None				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	DN Intention 1/9/61				
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:				
SPECIAL PROVISIONS	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.				
BY					
DATE:					

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:			
CUSTOMER Japan Air Lines	MCL 15,020 DTD			
CHANGE NO: 52	MODEL: 22-22 (Convair "880")			
TITLE: Specification Administrati Landing Gear Tail Skid)	ve Change (Revision to Auxiliary			
ORIGIN: Convair initiated				
REASON FOR CHANGE: Product improveme	n <b>t</b>			
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR, WT. EMPTY OPER, WT. EMPTY				
0 0	O INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *  None	,			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSED ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	SAL Anteuter 1/206.			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:				
DATE:	29 - (4 ( <sub>1</sub> - 4) <sub>1</sub> + 2 ) - 4			

Page 1 of 3

Title: Passenger Seating Configuration, Modification of

Origin: JAL Memo No. 26 (MITSUOKA to DUCKWORTH) dated 5 December 1960

Reason for Change: Customer request

Description of Change:

Page 1, Paragraph 1.1 BASIC TYPE:

Revise the third sentence as follows:

From: "The body of this specification describes an interior for a mixed seating arrangement which provides for the accommodation of 28 first class seats and 59 coach seats."

To: "This specification describes an interior seating arrangement which provides for the accommodation of 16 first class passengers and 74 coach passengers."

#### Page 3b, INTERIOR ARRANGEMENT:

Above illustration will be revised to show existing coat dividers immediately aft of fourth row of first class seats, and to show life rafts, stowed on floor, at a position forward of revised coat divider location. The revised illustration will also show three additional double coach seats and three additional triple coach seats in lieu of three rows of first class seats.

#### Page 6, Paragraph 3.1.2.2 TYPICAL LOADING SUMMARY:

Revise the below item under "Fixed Useful Load" as follows"

From: "Life Vests (105) 184 lb"

To: "Life Vests (97) 170 lb"

Revise the "Space Limit Payloads" as follows:

From:	"Space Limit Payloads:	Standard	-
	Passengers @ 165 1b (87 Standard) (104 Alternate Coad Baggage @ 40 lb/passenger Cargo	3,480 5,010	)
	91	otal 22,845	) ''

Page 2 of 3

To:	"Space Limits Payloads:	Standard
	Passengers @ 165 lb (90 Standard) (104 Alternate Coach)* Baggage @ 40 lb/passenger Cargo	14,850 3,600 4,890
	Total	23,340"

Page 82, Paragraph 3.19.1 GENERAL:

In the second line, change "87" to "90".

Page 89, Paragraph 3.19.3.1.1 PASSENGER COMPARTMENT:

Revise the first sentence as follows:

From: "The main passenger compartment aisle shall be a minimum of 24 inches wide between arm rests in the standard configuration."

To: "The main passenger compartment aisle width between arm rests shall be approximately 24 inches in the first class section and approximately 19 inches in the coach section."

Page A-2, APPENDIX I-B, CUSTOMER FURNISHED - CUSTOMER INSTALLED:

Revise the below item under "Description" as follows:

From: "105 Life Vest 184.0"
To: "97 Life Vest 170.0"

Page A-15, APPENDIX I-C, FURNISHINGS:

Revise the below items under "Description" as follows:

From: "7 Seats, Double, First Class, L-H 7 Seats, Double, First Class, R-H

12 Seats, Double, Coach, R-H
11 Seats, Triple, Coach, L-H
78 Tray, Integral Folding Food
87 Belt, Passenger Safety"

Page 3 of 3

" 4 Seats, Double, First Class, L-H To:

Seats, Double, First Class, R-H Seats, Double, Coach, R-H Seats, Triple, Coach, L-H Tray, Integral Folding Food Belt, Passenger Safety" 14 81 90

Effect on Weight Empty: Effect on Balance: Effect on Performance:

Effect on Operating Weight Empty: +26.0 pounds +40.0 pounds

-34,151 inch-pounds

None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO .: ZD- 22-322			DATE:			
CUSTOMER: Japan Air Lines		MCL 15,019 DTD				
		MODEL: 22-22 (Convair "880	11 )			
TITLE Specification Specification		(Revision to Engine				
ORIGIN: Convair initi	ated					
REASON FOR CHANGE: , T	REASON FOR CHANGE: , To specify the G.E. Engine Specification under which engines will be supplied.					
EFFECT	N WEIGHT *		EFFECT ON BALANCE *			
GUAR, WT. EMPTY	OPER. WT. EMPTY					
0	0		o INC	CH LB.		
EFFECT ON GUARANTEED PERFORMANCE: *  None						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Meadon 1/13/60			
			ENGINEERING APPRO	)VAL		
LATEST DATE OF ACCEPTANCE:		AIRPL	LANES AFFECTED:			
REC		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED: CONV		CONV	AIR, A DIVISION OF GENERAL DYNAMICS CO	RP.		
BY:						
DATE.			DON'V ALLE, 2	.D. 6-12174		

CONVAIR: SD

Specification Administrative Change (Revision to Engine Title:

Specification)

Origin: Convair initiated

Reason for Change: To specify the G.E. Engine Specification under

which engines will be supplied.

Description of Change:

Page 2, Paragraph 2.2 - ENGINE SPECIFICATION:

Revise the paragraph as follows:

"General Electric Engine Specification E-723h dated

1 September 1959 shall apply".

"General Electric Engine Specification E-790 dated To:

11 July 1960 with Reverser Specification E-725e dated 11 July 1960 and Suppressor Specification E-731f dated

11 July 1960 shall apply".

Page 4, Paragraph 3.1.1.1 - PERFORMANCE:

In the ninth line, change "E-723h, dated 1 September 1959" to "E-790, dated 11 July 1960".

Page A-3, APPENDIX I-C, PROPULSION EQUIPMENT:

Revise the following item under "POWER PLANT EQUIPMENT" as follows:

"4 Engine, complete \*(Dry) Gen. Electric CJ-805-3B" From:

"4 Engine, complete \*(Dry) Gen. Electric CJ-805-3B To:

Spec. No. E-790 Dated 7-11-60".

A UNISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322 DATE:			DATE:	
CUSTOMER: Japan Air	Japan Air Lines MCL 15,018 DTD		MCL 15,018 DTD	
CHANGE NO: 47		MODEL 22-22 (Convair "880		
	ation Administrati Language)	ve Ch	ange (Revision to Lighting	
ORIGIN: Convair : (Mitsuoka	initiated, based of to Duckworth) da	n Ite	m 4 of JAL Memorandum No. 26 December 1960	
REASON FOR CHANGE:	To more clearly dand aft lighting		e passenger compartment forward col.	
EFFECT	ON WEIGHT *		EFFECT ON BALANCE	
GUAR, WT. EMPTY	OPER, WT. EMPTY			
0	0		O INCH LB.	
EFFECT ON GUARANTEED P	ERFORMANCE: *			
	None	)		
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			Cluby 12/60 ENGINEERING APPROVAL	
		1		
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:		
STECIAL TROVISIONS.		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:		
ACCEPTED: - CONVAIR, A DIVISION OF GENERAL DYNAMICS CO		AIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:				
DATE			C14311	

CONVAIR: SD

Title: Specification Administrative Change (Revision to Lighting

Controls Language)

Origin: Convair initiated, based on Item 4 of JAL Memorandum No. 26

(Mitsuoka to Duckworth) dated 5 December 1960.

Reason for Change: To more clearly define passenger compartment for-

ward and aft lighting control.

Description of Change:

Page 63, Paragraph 3.16.8.3.3 - LIGHTING CONTROLS:

Add the following sentence to the end of the paragraph:

"The first class section lights shall be controlled from the forward cabin attendant's station, and the coach class section lights shall be controlled from the aft cabin attendant's station with the movable coat compartment dividers (Reference 3.19.3.1.2) located within the limitation of Stations 659 and 983 as defined in 3.20.1.9".

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:		
CUSTOMER: Japan Air Lines .			MCL 15017 DTD		
CHANGE NO: 46			MODEL: 22-22 (Convair "880"	')	
TITLE: Specific Cargo Ti	ation Administrat e-Down Fittings)	ive C	hange (Change to Locations o	f	
ORIGIN: Convair	ORIGIN: Convair initiated				
REASON FOR CHANGE: To more accurately define locations of cargo compartment tie-down fittings.					
EFFECT (	ON WEIGHT * *		EFFECT ON BALANCE*		
GUAR. WT. EMPTY	OPER. WT. EMPTY				
0	0		, O INCH	ILB.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N	alloqueum  ENGINEERING APPROV	/AL	
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:		
RE NO		RECUR NON-R	ET ON PRICE PER AIRPLANE: RRING: ECURRING:		
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORF		
BY:					
DATE:				6-12 = A	

Title: Specification Administrative Change (Change to Locations of

Cargo Tie-Down Fittings)

Origin: Convair initiated

Reason for Change: To more accurately define locations of cargo com-

partment tie-down fittings.

Description of Change:

Page 87, Paragraph 3.19.2.4 LUGGAGE AND CARGO COMPARTMENTS:

Delete all reference, in lower half of paragraph, to cargo tiedown fitting locations and substitute the following:

"FORWARD	COMPARTMENT	AFT COM	PARTMENT
B.L.	Station	B.L.	Station
32.32 L 16.50 L 10.50 R 32.38 L 16.50 R 32.38 L 10.50 R 32.38 L 10.50 R 16.50	388.50 388.50 388.50 388.50 388.54 425.44 425.44 425.44 425.47 463.50 463.50 514.47 514.47 5514.47 5519.58 589.58 589.58 589.58	27.22 I 16.50 R 27.22 R 27.22 R 26.50 R 27.24 I 16.50 R 27.24 I 16.50 R 25.65 R 26.82 R 26.82 R 26.82 R 26.82 R 26.82 R 26.82 R 26.82 R 26.82 R 26.82 R	1026.50 1026.50 1026.00 1054.00 1054.10 1054.10 1054.00 1092.00 1087.35 1092.10 1087.00 1140.00 1139.50 1144.90 1145.00 1178.60 1178.60 1178.60 1206.25 1206.25

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:		
CUSTOMER: Japan Air Lines	MCL 15,016 DTD		
CHANGE NO: 45A	MODEL: 22-22 (Convair "880")		
TITLE: Specification Administrative System Items)	e Change (Revision to Instrument		
ORIGIN: Convair initiated			
REASON FOR CHANGE: To provide current subject items, and	part numbers and description for revision to CCP No. 45.		
EFFECT ON WEIGHT *	EFFECT ON BALANCE*		
GUAR. WT. EMPTY OPER. WT. EMPTY			
0 0	O INCH L.B.		
EFFECT ON GUARANTEED PERFORMANCE: * None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	n ml Alantida		
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:		
	RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:			
DATE:	For various 4 4477		

Title: Specification Administrative Change (Revision to Instru-

ment System Items)

Origin: Convair initiated

Reason for Change: To provide current part numbers and description

for subject items, and revision to CCP No. 45.

Description of Change:

Page A-4, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below item under "SPERRY FLIGHT NAVIGATION SYSTEM" as follows:

From: "1 Vertical Gyro Sperry 617926-1"

To: "1 Vertical Gyro Sperry 2585297"

Delete the below item under "SPERRY C-11 REMOTE COMPASS SYSTEM":

"8 Amplifier, Servo Sperry 614937-51"

Page A-6, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "AUTO PILOT TYPE SP-30" as follows:

From: "1 Indicator, Auto Pilot Trim Sperry 1776004-1 1 Gyro Vertical Sperry 617926-1 1 Trim Servo Motor and Drive Sperry 1778879-41"

To: "1 Indicator, Auto Pilot Trim Sperry 1776004-5
1 Vertical Gyro Sperry 2585297
1 Trim Servo Motor and Drive Sperry 1778879-21"

\\ \int\_{i}^{2},

### CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO .: ZD- 22-322			DATE:	
CUSTOMER: Japan Air Lines			MCL 15,015 DTD	
CHANGE NO: 44			MODEL: 22-22 (Convair "880")	
TITLE Specification Control on A		Change	e (Indicate Marker Beacon	
ORIGIN: Convair init	iated			
	indicate marker leu of on pilots!		n control on ATC panel in ument panel.	
EFFECT	ON WEIGHT *		EFFECT ON BALANCE *	
GUAR. WT. EMPTY	OPER. WT. EMPTY		O INCH L.B.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N	Aleasty 1/21/60 ENGINEERING APPROVAL	
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:		
ACCEPTED: CONVAIR, A DIVISION OF GENERAL DYNAMICS CORT		NR, A DIVISION OF GENERAL DYNAMICS CORP.		
DATE:			F0974010.007-64.114	

CONVAIR: SD

Title: Specification Administrative Change (Indicate Marker Beacon

Control on ATC Panel)

Origin: Convair initiated

Reason for Change: To indicate marker beacon control on ATC panel

in lieu of on pilots' instrument panel.

Description of Change:

Page 79, Paragraph 3.17.3.2 - MARKER BEACON RECEIVER:

Revise the last sentence as follows:

From: "Controls for the marker beacon shall be located on the

pilots' instrument panel".

To: "Controls for the marker beacon shall be located on the

ATC control panel".

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:
CUSTOMER: Japan Air Lines		MCL 15,014 DTD	
CHANGE NO: 43			MODEL: 22-22 (Convair "880")
TITLE: Specification to Paragraph		hange	(Addition of Static Inverter
ORIGIN: Convair initi	ated		
REASON FOR CHANGE:	To specify that st	atic :	inverter is being furnished.
EFFECT (	ON WEIGHT *		EFFECT ON BALANCE *
GUAR, WT. EMPTY .	OPER, WT. EMPTY		1
0	0 0 .		O INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * None		
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N	Attenting 1/2/40 ENGINEERING APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:
R		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:	
ACCEPTED:			IR, A DIVISION OF GENERAL DYNAMICS CORP.
BY:			
DATE:			- Augustalia - Aug

CONVAIR: SD Japan Air Lines Change No. 43

Title: Specification Administrative Change (Addition of Static

Inverter to Paragraph 3.16.2.4.1)

Origin: Convair initiated.

Reason .for Change: To specify that static inverter is being furnished.

Description of Change:

Page 53, Paragraph 3.16.2.4.1 - POWER TRANSFER:

Add the following to the end of paragraph:

"A static inverter, connected to the battery, shall be provided for engine ground start when external a-c electrical power is not available".

4 DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:				
CUSTOMER: Japan Air Lines		MCL 15,013 DTD					
CHANGE NO: 42			MODEL:	55-55	(Convai	r "8	30")
TITLE: Specification Ceiling Heigh		nange	(Clarif	Cicati	on of Ca	bin	
	ated clarify cabin he:	ight 1	in the a	irea o	f the dr	oppe	i
EFFECT	N WEIGHT *			EFFE	CT ON BALA	ANCE .	
GUAR, WT. EMPTY O	OPER. WT. EMPTY O				0	ı	NCH LB.
EFFECT ON GUARANTEED F	EFFECT ON GUARANTEED PERFORMANCE: *  None						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:					dead ENGINEERIN		N/10/2
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFI	ECTED:			
SPECIAL PROVISIONS.		EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED: BY:		CONVA	IR, A DIVIS	ON OF G	ENERAL DYN	AMICS (	ORP.
DATE:						500V.10	V V-1

Title: Specification Administrative Change (Clarification of

Cabin Ceiling Height.

Origin: Convair initiated.

Reason for Change: To clarify cabin height in the area of the dropped

ceiling.

Description of Change:

Page 85, Paragraph 3.19.3.1.1 - PASSENGER COMPARTMENT:

Revise the last sentence to read as follows:

"The clear ceiling height in the main cabin shall be approximately 85 inches at the airplane centerline except in the area of the dropped ceiling which shall be approximately 78 inches".



A DIVISION OF GLNERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322		DATE:				
CUSTOMER: Japan Air Lines		MCL 15,012 DTD				
CHANGE NO: 41		MODEL: 22-22 (Convair 1880a)				
TITLE Specification Administrative Change 3.16.8.3.4 and to ADF Control Panel			(Revision to Paragraph			
ORIGIN: Convair init:	ORIGIN: Convair initiated					
REASON FOR CHANGE:	o clarify the inte	nt of	the specification			
EFFECT	N WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER, WT, EMPTY					
0	0		O INCH LB.			
EFFECT ON GUARANTEED P	ERFORMANCE: * Mone					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		Heady Q 1926 ENGINEERING APPROVAL				
LATEST DATE OF ACCEPTANCE:  AIRPI		AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS: EFFEC		ECT ON PRICE PER AIRPLANE:				
NON-RI		RING:ECURRING:				
ACCEPTED:		CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:						
DATE:			CONVANT, 5, 5, 6-12/14			

CO WAIR: SD

Specification Administrative Change (Revision to Paragraph 3.16.8.3.4 and to ADF Control Panel P/N)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification.

Description of Change:

Page 63, Paragraph 3.16.8.3.4 - READING LIGHTS:

Delete the last seven words in the fourth sentence, which reads as follows, and repunctuate sentence:

"and placarded to clearly indicate its function".

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "Control Panels" as follows:

From: "2 Radio Compass (ADF) Collins 614L-5"

614L-6" "2 Radio Compass (ADF) Collins illo:

ATE OF SECTION AND SECURITIONS OF SECTIONS

SPEC NO. ZD- 22-322			DATE:			
CUSTOMER Japan Al	r Lines		MCL 15,093 DTD 5 October 106			
CHANCE NO: 40			MODEL. 22-22 (Convair "886")			
TITLE Airtror	66430 Bidirectio	nal C	oupler, Installation of			
ORTERS JAL Me 1960	emo Mo. 11 (Mitsuo	ka to	Duckworth), dated 5 October			
REASON FOR CHANGE. CL	istomer request					
FFFECT	ON WEIGHT *		EFFECT ON BALANCE			
GHAR, NT. EMPTY	OPER, WT. EMPTY					
Wegligible	Negligible		Negligible INCHLB.			
LEFECT ON GUARANTEED P		one				
* NEGLIGIBLE CHANGES WILL TOTAL'S REFLECTED IN A ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL	Attacky 174/60 EINGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:			ANES AFFECTED			
SPECIAL PROVISIONS		EFFECT ON PRICE PER AIRPLANE RECURRING NON-RECURRING TOTAL:				
GERTER			AIR, A CIVISION OF GENERAL DYNAMICS CORP.			
48						
LAT F						

CONVAIR: SD

Title: Airtron 66430 Bidirectional Coupler, Installation of

Origin: JAL Memo No. 11 (Mitsuoka to Duckworth), dated 5 October

1960

Reason for Change: Customer request

Description of Change:

Page 80, Paragraph 3.17.4.1 WEATHER PENETRATION AIRBORNE RADAR:

Revise the last sentence as follows:

From: "Provisions shall be made in the waveguide installa-

tion for use of an Airtron 66430 Reflectometer."

Fo: "An Airtron 66430 bidirectional coupler shall be pro-

vided in the waveguide installation."

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Add the following item under "WEATHER RADAR":

"Coupler, Bidirectional Airtron 66430"

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None



A D. VISION OF GENERAL DYNAMICS CORPORATION SAN DIFFO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:	
CUSTOMER: Japan Ai	r Lines		MCL 15,092	DTD 26 September
CHANGE NO: 39			MODEL: 22-22	(Convair "880")
TITLE: Seat Paper	Dispenser, Instal	latio	n of	
	HARASAWA of JAL er 1960; and Memo			
EFFECT	N WEIGHT *		EFFECT	ON BALANCE *
GUAR, WT. EMPTY Negligible	OPER.WT.EMPTY Negligible		Negligible	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *	е		
* NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL N	galater	SINEERING APPROVAL
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:		RECUR NON-R	ET ON PRICE PER AIRI RRING: ECURRING:	
ACCEPTED:  BY:		CONVA	AIR, A DIVISION OF GENE	ERAL DYNAMICS CORP.

CONVAIR: SD

Japan Air Lines Change No. 39

Title: Seat Paper Dispenser, Installation of

Origin: Meeting, H. HARASAWA of JAL and H. B. DUCKWORTH of Convair

on 23 September 1960; and Memo from H. HARASAWA dated 23

September 1960

Reason for Change: Customer request

Description of Change:

Page 86, Paragraph 3.19.2.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Add the following item to the equipment list:

"One toilet seat paper dispenser"

Delete the below item from the equipment list:

"One toilet paper dispenser (interlock leaf)"

Page A-15, APPENDIX I-C, FURNISHINGS:

Add the following item under "Description":

"4 Dispensers, Toilet Seat Consol. Cover Co. PROCO"
Paper

Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None

The following shall not appear in the specification language:

The language effect of CCP No. 5, approved by Customer, is included in this proposal as concerns the toilet paper dispenser being deleted.



EAST OF GOLD AND FOR THE BAT OF

SPEC NO. ZD 22-322			DATE		
CUSTOMER Japan Air	Lines		MCL 15,011	DTD	
UH WHEE NO 37			MODEL: 22-22	(Convair	"880")
THEE Specification Language and	n Administrative Ch Filter P/N Change	nange )	(Revision to	Flare Di	spenser
ORGAN Convair init	iated				
SEASON FOR CHANGE. TO	clarify the inten	t of	the specifica	tion	
			EEEE	ET ON BALAN	ICE *
	ON WEIGHT .		F11 F	JI ON BALAI	
SUIGH, WI, EMPTY	OPER. WT. EMPTY		0		INCH LB.
O	1				
EFFECT ON GUARANTEED I	PERFORMANCE: '				
	Mone				
LIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL  COMPTANCE OF THIS CHANGE IS DEPENDENT UPON PRICE ACCEPTANCE OF THE FOLLOWING CHANGES.		AL		10/25/	
LATEST DATE OF ACCEPTA	ANCE:	AIRPL	ANES AFFECTED:		
SELCIAL PROVISIONS	•	RECUE NON-R	ET ON PRICE PER A RRING: ECURRING:		The second secon
A STILL		CONV	AIR, A DIVISION OF G	ENERAL DYNA	MICS CORP.

COFFIR: SD

Title: Specification Administrative Change (Revision to Flare

Dispenser Language and Filter P/N Change)

Origin: Convair initiated

Feason for Change: To clarify the intent of the specification

Description of Change:

Page 88, Paragraph 3.19.2.6.1 - FLARE DISPENSERS (PROVISIONS):

Delete "(PROVISIONS)" from paragraph title.

age 4-9, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "RADIO COMPASS (ADF) SYSTEM" as follows:

From: "2 Filters Collins 635-1"

To: "2 Filters Collins 635F-1"

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the specification language:

CCP No. 26 (Revision to Appendix I-C Part Numbers) inadvertently changed the ADF Filter part number from "635F-1" to "635-1". Approval of this proposal will restore the part number to original number which is correct.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:		
CUSTOMER: Japan Air I	ines		MCL 15,071	DTD 10 M	ay 1960
CHANGE NO: 36			MODEL: 22-22		
TITLE: True Air	speed Indicator, S	Blave	, Additional,	Installat	ion of
ORIGIN: JAL/Conv	air Meeting 10 May	1960	)		
REASON FOR CHANGE: 0	ustomer request				
			پر عمر مام مام مام مام	5 011 DAI 414	an par a
	N WEIGHT *		EFFEC	ON BALANO	JE *
GUAR. WT. EMPTY +3.0 lb	OPER. WT. EMPTY +3.0 lb			+810	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *	None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSA ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		AL N	a Avgheur 9/23/60		00
			EN	GINEERING	APPROVAL
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		RECUR NON-R	T ON PRICE PER AIF RING: ECURRING:	•	
ACCEPTED:		CONV	AIR, A DIVISION OF GEN	IERAL DYNAM	ICS CORP.
BY:					
DATE:				-e	WAS COMPANIED FOR

Title: True Airspeed Indicator, Slave, Additional, Installation of

Origin: JAL/Convair Meeting 10 May 1960

Reason for Change: Customer request

Description of Change:

Page 46, Paragraph 3.14.1.6 NAVIGATOR'S INSTRUMENTS:

Add the following item to the instrument list:

"One true airspeed indicator (slave)"

Page A-5, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Add the following item under "KOLLSMAN INTEGRATED FLIGHT INSTRU-MENT SYSTEM":

"1 Indicator, True Airspeed, Slave (Nav. Sta.)

Kollsman

A32355-10-001"

Effect on Weight Empty: +3.0 Pounds

Effect on Balance: +810 Inch-Pounds

Effect on Performance: None



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:	
CUSTOMER: Japan Air	Lines		MCL 15,010 DTD	
CHANGE NO: 35			MODEL: 22-22 (Convai	r "880")
Specificat 3.12.9.13		e Cha	nge (Revision to Para	graph
ORIGIN: Convair in	nitiated			
REASON FOR CHANGE: To	clarify the inte	nt of	the specification.	
EFFECT	N WEIGHT *		EFFECT ON BALA	ANCE *
GUAR, WT. EMPTY	OPER, WT. EMPTY			
0	0		0	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *			
	None			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		SAL N	a augueun 9/20/60	
			ENGINEERIN	G APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS:		RECUR NON-RI	ET ON PRICE PER AIRPLANE: RRING: ECURRING:	
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNA	AMICS CORP.
BY:				
DATE.				COLVERS CO. Account

CONVAIR: SD

Japan Air Lines Change No. 35

Title: Specification Administrative Change (Revision to Paragraph

3.12.9.13.2)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification.

Description of Change:

Page 41, Paragraph 3.12.9.13.2 EMERGENCY REFUELING:

In the third line, change the word "outboard" to "integral".

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:	
CUSTOMER: Japan Air	Lines		MCL 15,088 DTD 15	September :
CHANGE NO: 34			MODEL: 22-22 (Conve	air "880")
TITLE: VHF Navig	ation Instrument [	Jnits	, Change to	
ORIGIN: JAL Memor 15 Septem	andum No. 5 (S. M.	ITSUO	KA to H. B. DUCKWORTH	) dated
REASON FOR CHANGE:	Customer request			
				0
EFFECT O	N WEIGHT *		EFFECT ON BALA	NCE *
GUAR, WT. EMPTY	OPER, WT. EMPTY			
Negligible	Negligible		Negligible	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: *	ne		
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	Hatury 2) ENGINEERING	1) /4/40 G APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS:		RECUI NON-R	CT ON PRICE PER AIRPLANE: RRING: ECURRING:	
ACCEPTED:		CONV	AIR, A DIVISION OF GENERAL DYNA	AMICS CORP.
.BY:				
DATE:				NAMES OF

VHF Navigation Instrument Units, Change to Title:

JAL Memorandum No. 5 (S. MITSUOKA to H. B. DUCKWORTH) dated 15 September 1960

Reason for Change: Customer request

Description of Change:

Page A-9, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "VHF NAVIGATION SYSTEM" as follows:

"2 Instrument Units 344B-1" From: Collins

"2 Instrument Units Collins 344B-1A" To:

Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None

A DISTRICT OF GENERAL DYNAMICS CORPORATION
AND DISTRICT CORPORATION

SPEC NO. ZD- 22-322			DATE:	
CUSTOMER. Japan Air	r Lines		MCL 15,087	DTD 15 September
CHANGE NO. 33	-		MODEL: 22-22	(Convair "880")
TITLE Amplif:	ier, Speaker; Chan	ge to		W 10 10 10 10 10 10 10 10 10 10 10 10 10
ORIGIN JAL Mer 15 Sept	norandum No. 5 (S. cember 1960	MITS	UOKA to H. B.	DUCKWORTH) dated
REASON FOR CHANGE:	Customer request			
EFFECT	ON WEIGHT *		EFFEC	T ON BALANCE *
GUAR. WT. EMPTY	OPER. WT. EMPTY			
Negligible	Negligible		Neg	ligible INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * None			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPO ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES		SAL N	Menty	19 17/1/w
			EN	GINEERING APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS:		RECUR NON-RE TOTAL	ECURRING:	
ACCEPTED:				ERAL DYNAMICS CORP.
(1)				
CATS.				*** 14

Title: Amplifier, Speaker; Change to

Origin: JAL Memorandum No. 5 (S. MITSUOKA to H. B. DUCKWORTH)

dated 15 September 1960.

Reason for Change: Customer request

Description of Change:

Page A-9, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "P.A. SYSTEM" as follows:

From: "1 Amplifier (1/4 ATR) Collins 346D-1"

To: "1 Amplifier (1/4 ATR) Collins 346F-1"

Effect on Weight Empty: Negligible Effect on Balance: Negligible Effect on Performance: None

The following shall not appear in the specification language:

The 356F-1 amplifier will be bench set to a three-watt nominal output.

A 1 VIS ON OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

CHANGE NO: 32A MODEL: 22-22 (Convair "880  TITLE: Fuel Flow Totalizers, G. E., Installation of  ORIGIN: JAL representative request on 15 September 1960  REASON FOR CHANGE: Customer request  GUAR. WT. EMPTY OPER. WT. EMPTY  +48.0 lb +48.0 lb +25,345 INC  EFFECT ON GUARANTEED PERFORMANCE:	
TITLE: Fuel Flow Totalizers, G. E., Installation of  ORIGIN: JAL representative request on 15 September 1960  REASON FOR CHANGE: Customer request  EFFECT ON WEIGHT * EFFECT ON BALANCE *  GUAR. WT. EMPTY OPER. WT. EMPTY  +48.0 lb +48.0 lb +25,345 INC	")
ORIGIN: JAL representative request on 15 September 1960  REASON FOR CHANGE: Customer request  EFFECT ON WEIGHT * EFFECT ON BALANCE *  GUAR. WT. EMPTY OPER. WT. EMPTY +48.0 lb +25,345 INC	
REASON FOR CHANGE: Customer request  EFFECT ON WEIGHT * EFFECT ON BALANCE *  GUAR. WT. EMPTY OPER. WT. EMPTY +48.0 lb +25,345 INC	
EFFECT ON WEIGHT * EFFECT ON BALANCE *  GUAR. WT. EMPTY OPER. WT. EMPTY +48.0 1b +25,345 INC	
GUAR. WT. EMPTY OPER. WT. EMPTY +48.0 1b +25,345 INC	
+48.0 1b +48.0 1b +25,345 INC	
(2),519	
SECECT ON CHARANTEED DEPENDANCE +	HLB.
None None	
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL  ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:  ENGINEERING APPROV	VAL
LATEST DATE OF ACCEPTANCE:  AIRPLANES AFFECTED:	
SPECIAL PROVISIONS:  EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:	
ACCEPTED: CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP	Ρ.

CONVAIR: SI

Title: Fuel Flow Totalizer, G. E., Installation of

Origin: JAL representative request on 15 September 1960

Reason for Change: Customer request

Description of Change:

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS:

Above illustration will be revised to change "FUEL FLOW" indicator callout to "FUEL FLOW FUEL CONSUMED" and to add "FUEL RESET" switch.

Page 45, Paragraph 3.14.1.3 CENTER PANEL INSTRUMENTS:

Revise the below item in the instrument list as follows:

From: "\*Four fuel flow indicators"

To: "\*Four indicators, fuel flow/fuel consumed"

Page A-5, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below item under "ENGINE INSTRUMENTS - FUEL SYSTEM" as follows:

From: "4 Indicator, Fuel Flow Gen. Elec. 8DJ97LAA-1"

To: "4 Indicator, Fuel Flow, Gen. Elec. 8DJ77LAC-2"

Add the following items under the above item:

"4 Amplifier Gen. Elec. 8KE25GAC-1
4 Transmitter, Fuel Flow Gen. Elec. 8TJ59GAR3"
1 Static Inverter Electro

Devel. 2-107E"

Effect on Weight Empty: +48.0 pounds

Effect on Balance: +25,345 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIFFO, CALIFORNIA

3FEC NO. 20-22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,009 DTD			
CHANGE NO: 31			MODEL: 22-22 (Convair "880")			
TITLE: Specification Specification	n Administrative C n Language and Pil	hange ots i	e (Miscellaneous Revisions to Instrument Panel Illustration)			
ORIGIN: Convair initi	ated					
REASON FOR CHANGE: To	clarify the inte	nt of	f the specification			
	N WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY O	OPER. WT. EMPTY O		O INCH LB.			
EFFECT ON GUARANTEED P	ERFORMANCE: * None					
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF TI	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	alvagneur 9/14/60			
CCP No. 18			ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTAN	NCE:	AIRPL	LANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:						
DATE:			**************************************			

Page 1 of 2

Title: Specification Administrative Change (Miscellaneous Revisions to Specification Language and Pilots Instrument Panel Illustration)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification

Description of Change:

Page 11, TABLE OF CONTENTS:

Adjacent to 3.12.3, correct spelling of "Propulsion"

Page 25, Paragraph 3.8.4.7 - STEERING CONTROL:

In fifth line from bottom of paragraph, change "hear" to "gear".

Page 32, Paragraph 3.12.4.2.1 - CONSTANT SPRED DRIVE OIL SYSTEM:

In second line of first sentence, correct spelling of "engines".

Page 44, Paragraph 3.14.1 - INSTRUMENTS:

Delete the last six words in the first sentence "(without artificial light during daylight hours)", and repunctuate sentence.

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS:

Replace above illustration in the specification with Enclosure (A) Which relocated "VNE MNE WARNING TEST" and adds "RUDDER BOOST Q TEST & OVERRIDE" to pilot's panel.

Page 57, Paragraph 3.16.7.1.2 - FUSES:

Delete third sentence which is repetitious with second sentence.

Page 88, Paragraph 3.19.2.7.1 - ASH TRAYS:

Add the following to the ash tray list:

"One radio operator/navigator".

Page 88, Paragraph 3.19.2.7.7 - COAT STOWAGE:

Add the following after paragraph title, and delete the sentence:
"(Deleted)"

Japan Air Lines Change No. 31

Page 2 of 2

#### Page A-3, APPENDIX I-C:

In note on bottom of page correct spelling of "Including" and "reverse".

Page A-4, APPENDIX I-C:

Under "SPERRY C-11 REMOTE COMPASS SYSTEM", change "Flus" to "Flux"

Page A-18, APPENDIX I-D:

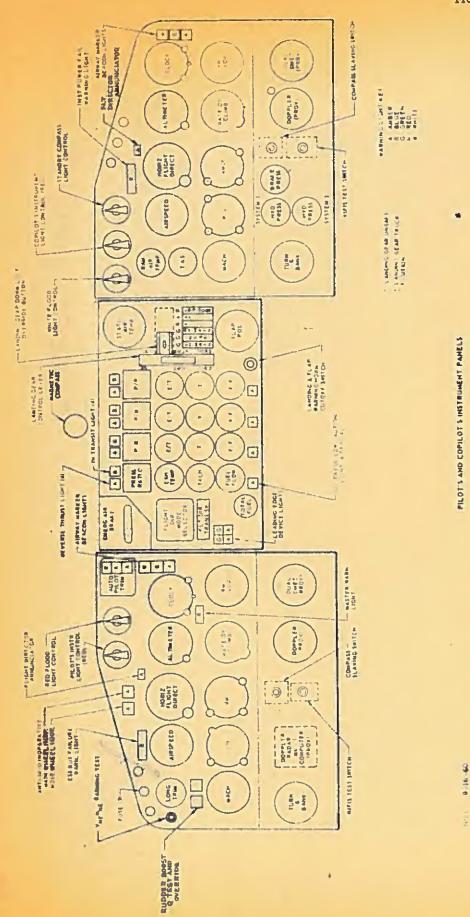
Under "Interior Trim", " change "Wanscot" to "Wainscot"

Enclosure: (A) One copy Page 44a - PILOT'S AND COPILOT'S INSTRUMENT PANELS (Revised)

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None

The following shall not appear in the specification language:

Enclosure (A) of this proposal includes the effects of CCP No. 7 (Flight Reference Selector Switch), approved by Customer; and the effects of CCP No. 18 (Revision to Pilots' Instrument Panel and Pedestal Illustrations). Therefore acceptance of this CCP is contingent on prior acceptance of CCP No. 18



A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIFFO, CALIFORNIA

SPEC NO.: ZD-22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,008 DTD			
CHANGE NO: 30			MODEL: 22-22 (Convair "880")			
TITLE: Specificat	tion Administrativ	re Cha	nge (Revised C. G. Limits Graph)			
ORIGIN: Convair in	nitiated		•			
REASON FOR CHANGE: TO	o furnish an up to	date	C. G. Limits Graph.			
	N WEIGHT *		EFFECT ON BALANCE *			
GUAR. WT. EMPTY	OPER. WT. EMPTY					
0	0		O INCH LB.			
EFFECT ON GUARANTEED P	ERFORMANCE: * None					
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
DATE:			- CONVARI, 5, D. 6-1217 A			

Title: Specification Administrative Change (Revised C. G. Limits

Graph)

Origin: Convair initiated

Reason for Change: To furnish an up to date C. G. Limits Graph.

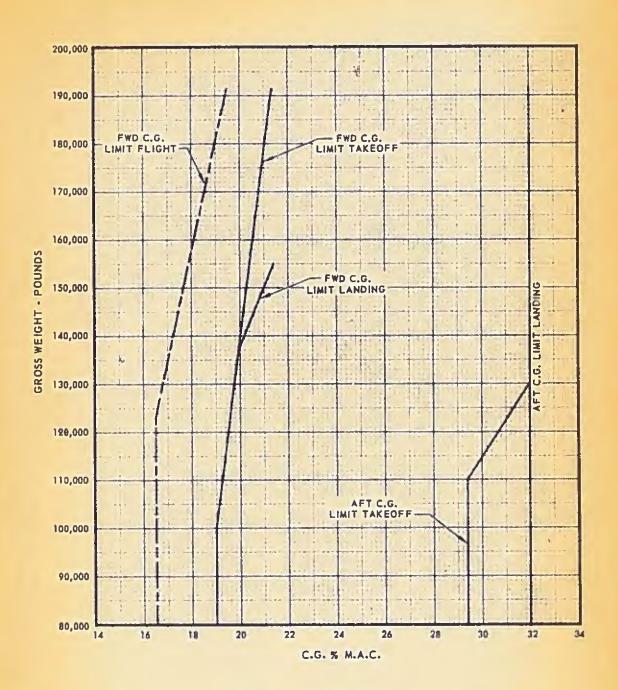
Description of Change:

Page 7a, C. G. DESIGN LIMITS:

Replace above illustration in the specification with Enclosure (A) which adds new aft C. G. Limit takeoff and landing.

Enclosure: (A) One copy Page 7a, C. G. LIMITS (Revised)

Effect on Weight Empty: 0
Effect on Balance: 0
Effect on Performance: None



C.G. LIMITS JAPAN

A DIVISIÓN OF GENERAL DYNAMICS CONFORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:					
CUSTOMER: Japan Air Lines			MCL 15,085 DTD 31 August 196			
CHANGE NO: 29			MODEL: 22-22	(Convai	· "880")	
TITLE ATC Transp	oonder Beacon, Dua	l, In	stallation of			
ORIGIN. JAL Memora 30 August	andum No. 3 (S. MI 1960	TUSOK	A to H. B. DU	CKWORTH)	, dated	
REASON FOR CHANGE: CI	ustomer request					
EFFECT	N WEIGHT *		EFFEC	T ON BALA	NCE *	
GUAR, WT. EMPTY	OPER. WT. EMPTY					
+50.0 lb	+50.0 lb		+13,350 INCH LB.			
EFFECT ON GUARANTEED P						
	None					
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL  ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:						
			E	NGINEERING	APPROVAL	
LATEST DATE OF ACCEPTANCE: -			AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED  BY:		CONVA	AIR, A DIVISION OF GE	NERAL DYNA	WICS CORP.	

Japan Air Lines Change No. 29

Page 1 of 2

Title: ATC Transponder Beacon, Dual, Installation of

Origin: JAL Memorandum No. 3 (S. MITUSOKA to H. B. DUCKWORTH), dated

30 August 1960

Reason for Change: Customer request

Description of Change:

Page 70, Paragraph 3.17.1 EQUIPMENT:

Revise the first sentence to read as follows:

"The following radio and radar systems shall be furnished and installed by Convair, except as otherwise noted."

Add the following item under "Description":

"\*ATC Transponder Beacon (Dual)"

Delete the following item under "Provisions for the following systems shall be made":

"ATC Transponder Beacon (Dual)"

Add the following to the bottom of Page 70:

"\*Buyer furnished equipment"

Page 80, Paragraph 3.17.4.2 RADAR SAFETY BEACON (PROVISIONS):

Revise the paragraph title and paragraph to read as follows:

"RADAR SAFETY BEACON: A Buyer-furnished dual, ATC Transponder Beacon shall be installed in the radio rack."

Page 80, Paragraph 3.17.4.2.2 ANTENNA:

Revise the sentence to read as follows:

"Two external-type antennas shall be installed."

Japan Air Lines Change No. 29

Page 2 of 2

Page 80a, ANTENNA ARRANGEMENT:

Above illustration will be revised to show two ATC Radar Beacon antennas installed in lieu of provisions for a second antenna.

Page A-1, APPENDIX I-A, CUSTOMER FURNISHED - CONVAIR INSTALLED:

Add the following to the page:

"ELECTRONIC EQUIPMENT

2 ATC Transponder Beacon Collins 621A-2 50.0 lb"

Page A-9, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "ATC TRANSPONDER BEACON" as follows:

"l Antenna Electronic LB-147" From:

"2 Antenna To: Electronic Specialties LB-147"

Effect on Weight Empty: +50.0 pounds

Effect on Balance: +13,350 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO, CALIFORNIA

SPEC NO. ZD- 22-322			DATE:	
CUSTOMER Japan Air Lines			MCL 15,006 DTD	
CHANGE NO: 27			MODEL: 22-22 (Convair "	880")
TITLE Specifica Disposal)	tion Administrativ	c Cha	ange (Delete One Towel	
ORIGIN Convair i	nitiated			
	To make specificat tory equipment ins		consistent with actual lation.	ava-
EFFECT	ON WEIGHT *		EFFECT ON BALANCE	4
GUAR, WT. EMPTY O O O O			0	INCH LB.
EFFECT ON GUARANTEED F	PERFORMANCE: *	ne		
NEGLIGIBLE CHANGES WIL TOTALS REFLECTED IN A ACCEPTANGE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPOR	SAL N	alloquenn 8/26/60	
•			ENGINEERING AF	PROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:	
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:		
ACCEPTED:		CONVA	AIR, A DIVISION OF GENERAL DYNAMIC	s CORP.
BY:				
DATE			ZONN	(A1-3.5.D. 6-1217

Title: Specification Administrative Change (Delete One Towel Dis-

posal)

Origin: Convair initiated

Reason for Change: To make specification consistent with actual lava-

tory equipment installation.

Description of Change:

Page 87, Paragraph 3.19.2.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Revise the top item on the page as follows:

From: "Two towel disposals (one for linen and one for paper)"

To: "One towel disposal"

Effect on Weight Empty: O Effect on Balance: O Effect on Performance: None

4 DEVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO. CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:	
CUSTOMER. Japan Air Lines	MCL 15,005 DTD	
CHANGE NO: 26	MODEL: 22-22 (Convair "880")	
TITLE Specification Administrative Equipment Part Numbers, and I	Change (Revision to Appendix I-C Delete Airline Shaver)	
ORIGIN: Convair initiated		
REASON FOR CHANGE: To provide current items, and JAL requbilingual list.	part numbers for subject equipment uest to delete Airline Shaver from	
EFFECT ON WEIGHT *	EFFECT ON BALANCE *	
GUAR, WT, EMPTY O O O	O INCH LB.	
EFFECT ON GUARANTEED PERFORMANCE: *	one	
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPORTION ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES Change Order No. 2	ON alogueurs 160	
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED: -	
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:	
ACCEPTED	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.	
BY		
DATE.	70.00	

Japan Air Lines Change' No. 26

Page 1 of 4

Title:

Specification Administrative Change (Revision to Appendix I-C

Equipment Part Numbers, and Delete Airline Shaver)

Origin:

Convair initiated

.Reason for Change:

To provide current part numbers for subject equip-

ment items, and JAL request to delete Airline Shaver

from bilingual list.

Description of Change:

Page 10, Paragraph 3.2.5 SIGNS, PLACARDS, NAMEPLATES:

Delete the below item under "The following signs shall be bilingual (Japanese-English)", and reletter the remaining items "x" to "aa":

"x. Airline Shaver"

Page A-4, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "FLIGHT INSTRUMENTS" as follows:

"1 From: Indicator, Longitudinal Trim U. S. Gauge SRL-012A Indicator, Flap Position (Dual) Gen. Electric 8DJ91LAB-Y-1 Indicator, Radio Magnetic Eclipse-Pioneer 36126-1AF-25-A-1 Indicator, Radio Magnetic"

"1 To: Indicator, Longitudinal Trim U. S. Gauge SRL-012C Indicator, Flap

Position (Dual) Gen. Elec. 8DJ91LAB-Y-1 or 8DJ91LAB-1 Indicator, Radio

Eclipse-Pioneer 36126-1AF-25-A-1" Add the following item under "FLIGHT INSTRUMENTS":

Magnetic

Servo Amplifier Collins 333B-3"

Under "SPERRY FLIGHT NAVIGATION SYSTEM", change the following:

"3 Servo Amplifier From: Sperry 618379"

Japan Air Lines Change No. 26

Page 2 of 4

"4 Servo Amplifier Sperry To: 618379"

Revise the below item under "SPERRY C-11 REMOTE COMPASS SYSTEM" as follows:

"2 From: Indicator, Radio Mag-

netic Direction

(with Power Failure Flag) (C-6B) Sperry"

To: Indicator, Radio Mag-

netic Direction (with Power Failure

Flag) (C-6B)

Sperry 1777215-623"

#### Page A-5, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Fill in part numbers for the below items under "ENGINE INSTRUMENTS-FUEL SYSTEM" as follows:

117	Indicator, Fuel Qty		
	Totalizer	Simmonds	393032-02929
2	Indicator, Fuel Qty Counter		
	Pointer Type, Tanks 1 & 4	Simmonds	383053-12919
2	Indicator, Refuel Quantity		
	Tanks 1 and 4	Simmonds	383093-04921
2	Indicator, Refuel Quantity		
	Tanks 2 and 3	Simmonds	383093-04922
5	Indicator, Fuel Qty Counter		
		Simmonds	383053-12920
1	Indicator, Fuel Quantity,		
	Center Section	Simmonds	383053-13930
1	Indicator, Refuel Quantity,		
	Center Section	Simmonds	383093-09420"

# Page A-6, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "ENGINE INSTRUMENT - OIL SYSTEM" as follows:

From:	1171	Indicator,	011	Quan-		
	1.	tity			Simmonds	393024-C1842
	4	Indicator,	Oil	Pres-		
	16	sure			U. S. Gauge	SR-04A
	4	Indicator,	011	Tem-		
		perature			Lewis	162023A"
m	HI.					
To:	"4	Indicator,	011	Quan-		
		tity			Simmonds	393024-03842

Japan Air Lines Change No. 26

Page 3 of 4

Indicator, Oil Pressure U. S. Gauge SR-04B 4 Indicator, Oil Temperature Lewis 162023"

# Page A-6, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "PRESSURIZATION, HEATING AND AIR CON-DITIONING" as follows:

From: "2 Indicator, Cabin Supercharger RPM Ham. Standard 535461 Indicator, Cabin Supercharger Bearing Temperature 800623" Fenwal To: Indicator, Cabin Ham. Standard Supercharger RPM 562208 Indicator, Cabin Supercharger Bearing Temperature Fenwa1 (P/N to be supplied)

# Page A-7, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "HYDRAULIC AND PNEUMATIC" as follows:

From:	"2	Indicator, Hydraulic		
		Pressure	U. S. Gauge	SRL-07J
	1	Indicator, Emergency Air Brake Pressure	II G G-	
	1	Indicator, Brake Hy-	U. S. Gauge	AW1826-AD01
	-	draulic Pressure	U. S. Gauge	SRL-07L
	2	Indicator, Hydraulic	o. b. dauge	PIM-OLF
		Fluid Temperature	Lewis	162C22A"
m	Ito			
To:	5	Indicator, Hydraulic		
		Pressure	U. S. Gauge	SRL-07K
	1	Indicator, Emergency		
	_	Air Brake Pressure	U. S. Gauge	AW1826-AD02
	1	Indicator, Brake Hy-		
	_	draulic Pressure	U. S. Gauge	SRL-07W
	2	Indicator, Hydraulic		
		Fluid Temperature	Lewis	162022"

Japan Air Lines Change No. 26

Page 4 of 4

Page A-9, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "RADIO COMPASS (ADF) SYSTEM" as follows:

From: "2 Filters Collins 635F-1"

To: "2 Filters Collins 635-1"

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "HF COMMUNICATION" as follows:

From: "1 Frequency Chart Gables G-587"

To: "1 Frequency Chart Pan-American PA-13C"

Revise the below item under "CONTROL PANELS" as follows:

From: "1 Dual IIF Control Panel Gables G-1033"

To: "1 Dual HF Control Panel Gables G-1033V"

Page A-11, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "LORAN" as follows:

From: "1 Receiver EDO 22760"

To: "1 Receiver EDO 31742"

Effect on Weight Empty: O
Effect on Balance: O

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE 40, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:		
CUSTOMER Japan Air Lines	MCL 15,084 DTD 11 August 1960		
CHANGE NO: 25	MODEL: 22-22 (Convair "880")		
TITLE: Smoke Goggles, Change in	Stowage of		
ORICIN: Japan Air Lines requested			
REASON FOR CHANGE: As above			
EFFECT ON WEIGHT *	EFFECT ON BALANCE *		
GUAR. WT. EMPTY OPER. WT. EMPTY			
Neg. Neg.	Neg. INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: * Non			
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROP ACCEPTANCE OF THIS CHANGE IS DEPENDENT UP PRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	OSAL a Chagerens 81-1		
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:		
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:			
DATE:	CONVAIL-CLUSTAL)		

Japan Air Lines Change No. 25

CONVAIR: SD

Smoke Goggles, Change in Stowage of Title:

Origin: Japan Air Lines requested

Reason for Change: As above

Description of Change:

Page 45a, PILOT'S AND COPILOT'S CONSOLE AND WING PANELS:

Above illustration will be revised to delete "SMOKE GOGGLES" in four places.

Page 93, Paragraph 3.19.5.5 MASKS:

Add the following to the end of the paragraph:

"Hooks shall be provided in the pilot compartment for stowage of five crew smoke goggles."

Effect on Weight Empty: Negligible Negligible Effect on Balance:

Effect on Performance: None

1 11 ISION OF GENERAL DYNAMICS CORPORATION SAN JIL 20, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:					
CUSTOMER Japan Air Lines	MCL 15,084 DTD 17 October 196					
CHANGE NO. 25A	MODEL: 22-22 (Convair "880")					
TITLE Specification Administrative Goggles)	Change (Change in Stowage of Smoke					
ORIGIN. Convair initiated	•					
REASON FOR CHANGE. To clarify smoke goggle stowage, and revision to CCP No. 25.						
EFFECT ON WEIGHT *	EFFECT ON BALANCE					
GUAR, WT. EMPTY OPER. WT. EMPTY						
Negligible Negligible	Negligible INCH L8.					
EFFECT ON GUARANTEED PERFORMANCE: * None						
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	ENGINEERING APPROVAL					
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.					
BY.						
DATE						

Japan Air Lines Change No. 25A

CONVAIR: SD

Specification Administrative Change (Change in Stowage of Title:

Smoke Goggles)

Convair initiated Origin:

To clarify smoke goggle stowage, and revision to CCP No. 25 Reason for Change:

Description of Change:

Page 45a, PILOT'S AND COPILOT'S CONSOLE AND WING PANELS:

Above illustration will be revised to delete "SMOKE GOGGLES" in four places.

Page 93, Paragraph 3.19.5.5 MASKS:

Add the following to the end of the paragraph:

"Holders shall be provided in the pilot compartment for the stowage of five crew smoke goggles."

Effect on Weight Empty: Negligible Negligible Effect on Balance: Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322	DATE:			
CUSTOMER: Japan Air Lines	MCL 15,068 DTD 10 May 1960			
CHANGE NO: 24	MODEL: 22-22 (Convair "880")			
TITLE: Audio System, Changes to				
ORIGIN: JAL/CONVAIR Meeting 10 M	ay 1960			
REASON FOR CHANGE: Customer request	•			
EFFECT ON WEIGHT *	EFFECT ON BALANCE *			
GUAR. WT. EMPTY OPER. WT. EMPTY				
-12.0 lb / -12.0 lb	-3,114 INCH LB.			
EFFECT ON GUARANTEED PERFORMANCE: *	ne			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROPORTION ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGE:	ON Chaqueum [10]			
- FRIOR ACCEPTANCE OF THE POLLOWING CHANGE	ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:			
	RECURRING:			
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:				
DATE:	CON VAIR, S.D. 6-1217A			

1:

Japan Air Lines Change No. 24

Page 1 of 2

Title: Audio System, Changes to

Origin: JAL/CONVAIR Meeting 10 May 1960

Reason for Change: Customer request

Description of Change:

Page 44b, PILOT'S AND COPILOT'S PEDESTAL:

Above illustration will be revised to delete "PUBLIC ADDRESS" and show "SPACE" in lieu of.

Page 45a, PILOT'S AND COPILOT'S CONSOLE AND WING PANELS:

Above illustration will be revised to delete pilot's and copilot's "MIKE SELECT PANEL".

Page 45b, FLIGHT ENGINEER'S PANEL:

Above illustration will be revised to delete "MIKE SELECT".

Page 45c, PILOT'S AND COPILOT'S OVERHEAD CONTROL PANEL:

Above illustration will be revised to show "PUBLIC ADDRESS".

Page 70, Paragraph 3.17.1.1 CONTROL PANELS:

Revise the below item under "Description" as follows:

From: "Audio Selector (5)"

To: "Audio Selector (4)"

Delete the following item:

"Microphone Selector (4)"

Page 76, Paragraph 3.17.2.3.1 AUDIO SELECTOR:

Delete the third sentence which reads: "An audio selector panel shall be provided near the radio rack."

1

Japan Air Lines Change No. 24

Page 2.of 2

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "CONTROL PANELS" as follows:

From: "5 Audio Selector Panel Gables G-567"

To: "4 Audio Selector Panel Pan-Air Electronics PA-41G"

Delete the below item under "CONTROL PANELS":

"4 Microphone Selector Panel Gables G-802"

Effect on Weight Empty: -12.0 pounds

Effect on Balance: -3,114 inch-pounds

Effect on Performance: None

The following shall not appear in the specification language:

This proposal also includes the installation of the following:

- (a) Three, 2-PDT muting relays on audio junction panel.
- (b) Larger version Frequency Chart.

A TIVENON OF FENERAL DYNAMICS CORPORATION SAN LIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:		
CUSTOMER: Japan Air Lines			MCL 15.076	TD 20 M	ay 1960
CHANGE NO: 23			MODEL: 22-22 (Convair "880")		
TITLE Squelch Contro	ol on VHF Communic				
	106	^			
ORIGIN. JAL/Convair M	eeting, 19 May 190	U			
REASON FOR CHANGE: C	ustomer request				
,					
EFFECT	N WEIGHT .		EFFECT	ON BALANC	CE '
GUAR, WT. EMPTY	OPER. WT. EMPTY		Neg.		
Neg.	Neg.		neg.		INCH LB.
EFFECT ON GUARANTEED F	ERFORMANCE. None				
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS	4	alugnen	n 8/10	160
			ENG	GINEERING	APPROVAL
LATEST DATE OF ACCEPTA	NCE:	AIRPL	ANES AFFECTED:		
SPECIAL PROVISIONS.		EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED.		CONV	AIR, A DIVISION OF GEN	ERAL DYNAM	AICS CORP.
ВҮ					
DATE.					19 M 12 1/24

CONVAIR: SD Japan Air Lines Change No. 23

Title: Squelch Control on VHF Communication Control Panel,

Installation of

Origin: JAL-CONVAIR Meeting, 19 May 1960

Reason for Change: Customer request

Description of Change:

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Revise the below item under "CONTROL PANELS" as follows:

From: "VHF Comm/VHF Nav. (ILS, DMET) Gables G-510V"

To: "VHF Comm/VHF Nav. (ILS, DMET) Gables (P/N to be supplied)"

Effect on Weight Empty: Neg. Effect on Balance: Neg. Effect on Performance: None

The following shall not appear in the specification language:

The new VHF Communication Control Panel will be similar to the panel it replaces except a squelch control feature will be incorporated therein.

A IH. ISLN OF GENERAL DYNAMICS CORPORATION SAN DIF GO, CALIFORNIA

SPEC ND.: ZD- 22-322		DATE:			
CUSTOMER: Japan Ai	r Lines	I	MCL 15.075	DTD 20 N	lay 1960
CHANGE NO 22		1	MODEL: 22-22	(Convair	"880")
TITLE Two C-11 of	. Compass Controlle	ers on	Overhead Pane	el, Instal	lation.
ORIGIN. JAL/COM	VAIR Meeting 19 Mag	1960			
REASON FOR CHANGE:	Customer request				
EFFECT (	ON WEIGHT *		EFFECT	ON BALANC	E *
GUAH, WT. EMPTY	OPER, WT. EMPTY				
+8.0 lb	+8.0 lb			+2,197	INCH LB.
EFFECT ON GUARANTEED F	PERFORMANCE: * None	9			
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CH PRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	a avgere	ur 8/10	
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANÉ: RECURRING: NON-RECURRING: TOTAL:			
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY·					
DATE:				- 5 W N	41 1 3 3 4 13 14

Japan Air Lines Change No. 22

Two C-11 Compass Controllers on Overhead Panel, Installation Title:

of

JAL/CONVAIR Meeting 19 May 1956

Reason for Change: Customer request

Description of Change:

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS:

Above illustration will be revised to delete the compass slaving switch from the pilot's to copilot's panels.

Page 45c, PILOT'S AND COPILOT'S OVERHEAD CONTROL PANEL:

Above illustration will be revised to show two C-11 compass controllers.

Page A-4, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Add the following items under "SPERRY C-11 REMOTE COMPASS SYTEM:

Compass Controllers Sperry 1775132 Power Supply Modules Sperry 1775143"

Effect on Weight Empty: +8.0 pounds Effect on Balance: +2,197 inch-+2,197 inch-pounds

Effect on Performance: None

A DIVITION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO. ZD- 22-322			DATE:		
CUSTOMER: Japan Air Lines			MCL 15,073 DTD 20 May 1960		
CHANGE NO 21			MODEL: 22-22 (	Convair	"880")
THILE: Dual Control	Panel, ATC Transpo	nder,	Installation	of	
ORIGIN JAL/CONVAIR	Meeting, 19 May 19	60			
REASON FOR CHANGE: C	ustomer request				
EFFECT ON WEIGHT *		EFFECT ON BALANCE			
GUAS, WT. EMPTY Jeg.	OPER. WT. EMPTY		Ne	g.	INCH LB.
EFFECT ON GUARANTEED P	ERFORMANCE: * Hone				
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:		١		NGINEERIN	G APPROVAL
LATEST DATE OF ACCEPTANCE:			ANES AFFECTED:		
SPECIAL PROVISIONS.		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.			
ВҮ					
[ATL:			-		En ( kir 👵 ( ) 🦏 ()

Japan Air Lines Change No. 21

CONVAIR: SD

Title: Dual Control Panel, ATC Transponder, Installation of

Origin: JAL, CONVAIR Meeting, 19 May 1960

Reason for Change: Customer request

Description of Change:

Page 71, Paragraph 3.17.1.1 - CONTROL PANELS:

Add the following item under "Description":

"ATC Transponder Beacon (Dual)"

Delete the below item under "Provisions for the following control panels shall be made:"

"ATC Transponder Beacon (Dual)"

rage 60, Paragraph 3.17.4.2.1 - CONTROL:

Revise the sentence to read as follows:

"One dual ATC Transponder Beacon control panel shall be installed on the Lilot's pedestal."

Page A-10, APPENDIX I-C, ELECTRONIC EQUIPMENT:

Add the following item under "CONTROL PANELS"

"Dual ATC Transponder Gables G-565"

Elice on Meight Empty: Meg. Elice on Miance: Neg. Elice on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

CUSTOMER: Japan Air Lines  CHANGE NO: 20A			MCL 15,077 DTD 23 August 19				
		MODEL: 22-22 (Convair "880")					
TITLE: Autopilot Part Numb	Flight Control Coners of	mputer	r and Controller	, Revisi	on to		
ORIGIN: JAL/Conva	ir Meeting of 19 M	ay 196	50				
REASON FOR CHANGE:	Customer request,	and re	evision to CCP N	lo. 20.			
		٤					
EFFECT ON WEIGHT *		EFFECT ON BALANCE *					
GUAR, WT. EMPTY	OPER. WT. EMPTY	-	N				
+1.0 lb	+1.0 lb			+300	INCH LB.		
EFFECT ON GUARANTEED	PERFORMANCE: * None	1					
ACCEPTANCE OF THIS CI	LL BE ACCUMULATED AND A FUTURE CHANGE PROPOS HANGE IS DEPENDENT UPO THE FOLLOWING CHANGES	SAL N	uph -	ndcoc			
			ENG	NEERING AF	PPROVAL		
LATEST DATE OF ACCEPTANCE:		AIRPLANES AFFECTED:					
		EFFEC	T ON PRICE PER AIRP	LANE:			
		RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:		CONVA	AIR, A DIVISION OF GENER	RAL DYNAMIC	s CORP.		
BY:	~						
DATE:			r	SONV	AIH, S.D. 6-1217		
	1						

Japan Air Lines Change No. 20A

Title: Autopilot Flight Control Computer and Controller, Revision to

Part Numbers of

Origin: JAL/Convair Meeting of 19 May 1960

Reason for Change: Customer request, and revision to CCP No. 20.

Description of Change:

Page A-6, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Revise the below items under "AUTOPILOT TYPE SP-30" as follows:

From: "1 Computer, Flight Control Sperry 1776003-1 1 Automatic Pilot Controller Sperry 1776001-1"

To: "1 Computer, Flight Control Sperry (P/N to be suppli-1 Automatic Pilot Controller Sperry 1776001-5"

Effect on Weight Empty: +1.0 pound

Effect on Balance: +300 inch-pounds

Effect on Performance: None

The following shall not appear in the specification language:

The computer will be similar to Sperry 1776003-1 except that it will be modified to include provisions for doppler input.

This proposal also includes the installation of wiring between the autopilot flight control computer, doppler navigator computer and doppler controller.



CONVAIR

A DIVISION OF GENERAL DYNAMICS CORPORATION
SAN DIF GO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:		
CUSTOMER Japan Air Lines	MCL 15.004 DTD		
CHANGE NO: 18	MODEL: 22-22 (Convair "880")		
TITLE: Specification Administrative Instrument Panel and Pedestal	Change (Revision to Pilots' Illustrations)		
ORIGIN. Convair initiated			
REASON FOR CHANGE: To clarify the int	cent of the specification		
EF, FECT ON WEIGHT *	EFFECT ON BALANCE *		
GUAR, WT. EMPTY OPER, WT. EMPTY			
Neg. Neg.	Neg. INCH LB.		
EFFECT ON GUARANTEED PERFORMANCE: *			
None			
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSED ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES	DN allegerens 8/2/60		
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:		
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:		
ACCEPTED	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY:			
DATE:	-08 9 km (2.00 km) 17 km		

CONVAIR: SD Japan Air Lines Change No. 18

Title: Specification Administrative Change (Revision to Pilots!

Instrument Panel and Pedestal Illustrations)

Origin: Convair initiated

Reason for Change: To clarify the intent of the specification.

Description of Change:

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS:

Replace above illustration in the specification with Enclosure (A) which deletes "MAIN LANDING GEAR" and "NOSE LANDING GEAR" on pilot's panel and adds "MAIN WHEEL INOP."

Page 44b, PILOT'S AND COPILOT'S PEDESTAL:

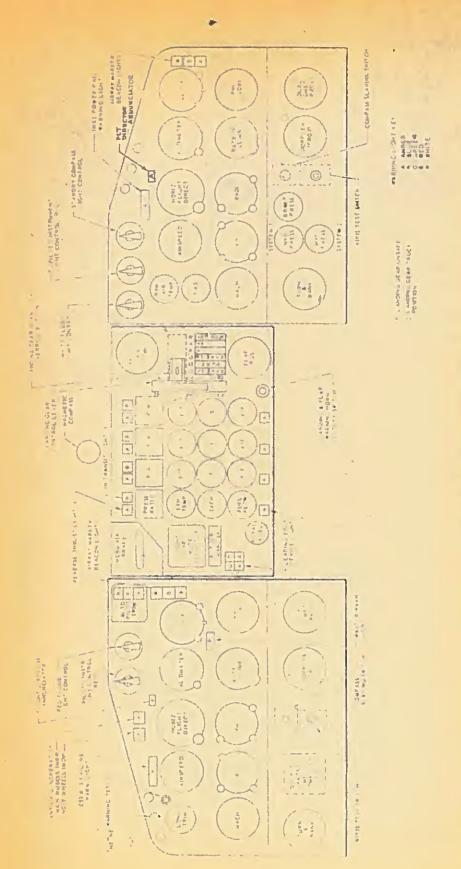
Replace above illustration in the specification with Enclosure (B), which deletes "MALFUNCTION LIGHTS".

Enclosures: (A) One copy Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS (Revised)

(B) One copy Page 44b, PILOT'S AND COPILOT'S PEDESTAL (Revised)

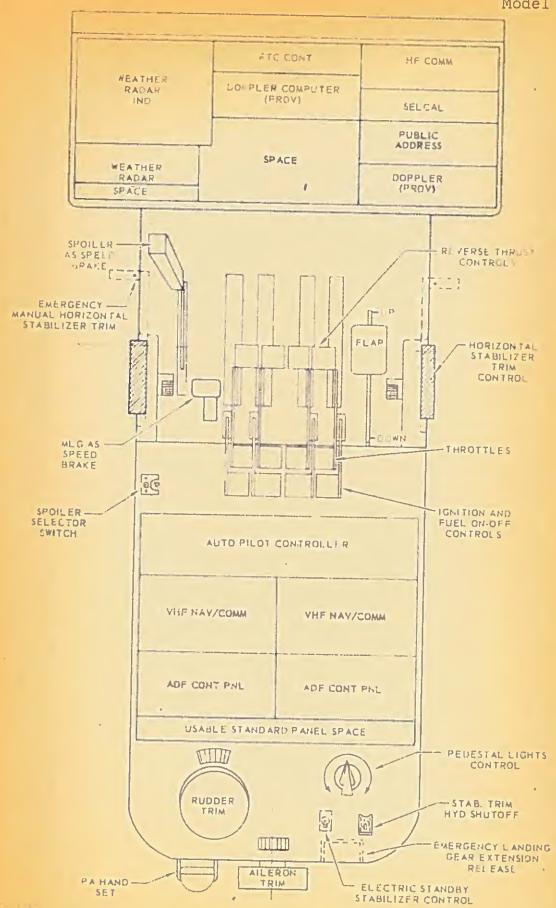
Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None



PILOT S AND COPILOT S INSTRUMENT PAHELS

Report No. ZD-22-321 Model 22M



PILOT'S AND CO-PILOT'S PEDESTAL

A DIVISION OF GENERAL DYNAMICS CORPORATION - SAN DIEGO, CALIFORNIA

SPECINO.: ZD- 22-322			DATE:		
CUSTOMER: Japan Air Lines		MCL 15,083 DTD			
CHANGE NO: 17			MODEL: 22-22 (Convair "880")		
TITLE: Main Syste	em Hydraulic Pumps	S, Cha	ange to		
ORIGIN Convair in	nitiated				
REASON FOR CHANGE: Pro	oduct improvement				
ЕГРЕСТО	N WEIGHT *		EFFECT ON BALANCE		
GUAF. WT. EMPTY	OPER. WT. EMPTY				
+31.0 1b	+31.0 lb		+23,091 INCH LB.		
EFFECT ON GUARANTEED P	ERFORMANCE: *				
	None				
NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A R ACCEPTANCE OF THIS CHA PRIOR ACCEPTANCE OF TH	FUTURE CHANGE PROPOS NGE IS DEPENDENT UPOR	AL 4	alloqueum 7/20/60  ENGINEERING APPROVAL		
LATEST DATE OF ACCEPTAN	ICF-	AIRPI	ANES AFFECTED:		
ENTERT DATE OF ACCEL TAI		7.11.12.	ANTEO ATTESTED,		
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:			
ACCEPTED:			IR, A DIVISION OF GENERAL DYNAMICS CORP.		
BY.					
TATE					

COMVAIR: SD Japan Air Lines Change No. 17

Title: Main System Hydraulic Pumps, Change to

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page A-12, APPENDIX I-C, HYDRAULIC AND PNEUMATIC EQUIPMENT:

Revise the below item under "Description" as follows:

From: "4 Pump, Main System

(Variable Displacement

Type) Ham. Standard"

To: "4 Pump, Main System

(Variable Displacement

Type) Vickers AS-61695-L-2"

Effect on Weight Empty: +31.0 pounds

Effect on Balance: +23,091 inch-pounds

Effect on Performance: None

The following shall not appear in the Specification language:

The hydraulic pump will incorporate one electrical depressurization valve and one blocking valve. The bolt-on ports will be modified to suit the Model 22 installation, and the yoke blocked back to a 6 gpm rating in lieu of 10 gpm.

This proposal also deletes the four accumulators in the hydraulic return lines, since these accumulators are not required with the Vickers pump.

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:					
CUSTOMER: Japan Air Lines			MCL _	15,003	_DTD	24	June	1960
CHANGE NO: 14			MODEL	: 22-22	(Conv	air	"880	")
TiTLE: Specific Provision		nange	(Signal	Pistol	Sto	wage		
ORIGIN: Convair	initiated		(u ),					
	o relocate for sa assenger access to				revent	ing		
EFFECT O	ON WEIGHT *			EFFE	CT ON BAL	ANC	E *	
GUAR, WT. EMPTY	OPER, WT, EMPTY							
Neg.	Neg.		Neg. INCH LB				LB.	
EFFECT ON GUARANTEED P	PERFORMANCE: * None							
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL	61	ivgerens 28/6°	NGINEERII	NG AI	PPROV	<b>AL</b>
LATEST DATE OF ACCEPTA	NCE:	AIRPLA	ANES AF	FECTED:				
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:						
ACCEPTED:			IR, A DIV	ISION OF GE	VERAL DYN	IAMIC	S CORP.	
DATE:						Signa ti	41 -, 5, 6,	0-1_17 A

CONVAIR: SD Japan Airlines Change No. 14

Title: Specification Administrative Change (Signal Pistol Stowage

Provisions, Relocation of)

Origin: Convair initiated

Reason for Change: To relocate for safety precautions, preventing

passenger access to these items.

Description of Change:

Page 88, Paragraph 3.19.2.6.2 SIGNAL PISTOL:

Change paragraph and title to read as follows:

"SIGNAL PISTOL (PROVISIONS): Provisions for the stowage of one Kilgore G-37-K signal pistol and six Kilgore 3-27 shells shall be made in the pilot's compartment. These items shall be Customer furnished.

Effect on Weight Empty: Negligible Effect on Balance: Negligible

Effect on Performance: None

A UIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO. ZD- 22-322	DATE:					
CUSTOMER. Japan Air Lines	MCL 15,074 DTD 20 May 1960					
CHANGE NO: 12	MODEL: 22-22 (Convair "880")					
TITLE: Spare Radio Equipment Stowage in Electronic Rack, Provisions for						
ORIGIN: JAL/CONVAIR Meeting of 19 Ma	ay 1960					
REASON FOR CHANGE: Customer request						
EFFECT ON WEIGHT *	EFFECT ON BALANCE *					
GUAR. WT. EMPTY OPER. WT. EMPTY +2.0 lb	+550 INCH LB.					
EFFECT ON GUARANTEED PERFORMANCE: * None						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A RUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES.	SAL allegueur 7/3/60					
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED  BY:  DATE:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.					

19

Title: Spare Radio Equipment Stowage in Electronic Rack, Provisions

for

Origin: JAL/CONVAIR Meeting of 19 May 1960

Reason for Change: Customer request

Description of Change:

Page 80, Add the following new paragraph after Paragraph 3.17.3.7.2:

"3.17.3.7.3 SPARE RADIO EQUIPMENT STOWAGE PROVISIONS: Stowage provisions (less wiring) shall be made in the electronic rack for the following spare radio equipment:

> 1 H.F. Transmitter, ATOM (1/2 ATR) Α.

C.

1 H.F. Receiver, STAR (1/2 ATR)
1 ADF Receiver, 51Y-3 (1/2 ATR)
1 Instrument Unit, 344B-1 (1/2 ATR)
1 Loran Receiver, 31742 (3/4 ATR)" D.

E.

Effect on Weight Empty: +2.0 pounds

Effect on Balance: +550 inch-pounds

Effect on Performance: None



A 14 ISION OF GENERAL DYNAMICS CORPORATION SAN DIF GO, CALLFORNIA

SPEC NO.: 20- 22-32	2		DATE		
CUSTOMER: Japan Air Lines			MCL 15,002	DTD <u>15</u> J	une 1960
CHANGE NO- 11			MODEL: 22-22	(Convair	"880")
Relocation	on Administrative Ch	nange	(Emergency Rad	iio Trans	mitter,
ORIGIN Convair ini	tiated				
	To correct actual loin the specification		n of Emergenc	y Radio T	ransmitte
EFFEC	T ON WEIGHT .		EFFEC	T ON BALANO	E *
GUA . WT. EMPTY	OPER, WT. EMPTY				
None	None		N	one	INCH LB.
EFFECT ON GUARANTEE	PERFORMANCE: * None				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			alwynenn	6/11/6	v
			EN	IGINEERING A	APPROVAL
LATEST DATE OF ACCEP	TANCE:	AIRPL	ANES AFFECTED:	77	
. RE		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:			
		JATCT	:	1,400	
ACCEPTED		CONVA	IR, A DIVISION OF GEN	IERAL DYNAM	C\$ CORP.
ВҮ				4	
DATE:					4.8 0,027,00

Japan Air Lines Change No. 11

CONVAIR: SD

Title: Specification Administrative Change (Emergency Radio Transmitter,

Relocation of)

Origin: Convair initiated

Reason for Change: To correct actual location of Emergency Radio Trans-

mitter in the Specification.

Description of Change:

Page 80, Paragraph 3.17.3.6 - EMERGENCY RADIO TRANSMITTER:

Change the paragraph to read as follows:

"Provisions shall be made for the stowage of one AN/CRT-3 emergency radio transmitter in the right hand forward coat compartment".

Effect on Weight Empty: None Effect on Balance: None Effect on Performance:

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:					
CUSTOMER: Japan Air Lines	MCL 15,078 DTD 8 June 1960					
CHANGE NO: 10	MODEL: 22-22 (Convair "880")					
TITLE Main Tires, Fabric Reinforced, Installation of						
ORIGIN: Convair initiated						
REASON FOR CHANGE: Product improvement	•					
EFFECT ON WEIGHT *	EFFECT ON BALANCE *					
GUAR, WT. EMPTY OPER, WT. EMPTY						
+23.0 lb +23.0 lb	+20,194 INCH LB.					
EFFECT ON GUARANTEED PERFORMANCE: *  Non	le .					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AN TOTALS REFLECTED IN A FUTURE CHANGE PROPORTION ACCEPTANCE OF THIS CHANGE IS DEPENDENT UP PRIOR ACCEPTANCE OF THE FOLLOWING CHANGE	ON S:					
	ENGINEERING APPROVAL					
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:					
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.					
BY:						
DATE:	PONVATION					

Japan Air Lines Change No. 10

CONVAIR: SD

Title: Main Tires, Fabric Reinforced, Installation

Origin: Convair initiated

Reason for Change: Product improvement

Description of Change:

Page A-14, APPENDIX I-C, LANDING GEAR EQUIPMENT:

Revise the below item under "Description" as follows:

From: "8 Tire, Main Wheel (39 x 13)
Type VII 22 Ply Rating,
Skid Depth .25-inches)
U. S. Rubber"

To: "8 Tire, Main Wheel (39 x 13)
(Type VII, 22-Ply Rating,
Fabric-Reinforced Tread,
Skid Depth .35-inch)
U. S. Rubber"

Effect on Weight Empty: +23.0 pounds

Effect on Balance: +20,194 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS COMPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:		11-13pt	
CUSTOMER: Japan Air Lines			MCL 15,070 DTD 10 May 1960			
CHANGE NO: 9			MODEL: 22-22	(Convair	"880")	
TITLE: Radio and Flight Director Sys			Interconnec	tion of		
ORIGIN: JAL/CONVAI	R Meeting of 10 Ma	y 196	0			
REASON FOR CHANGE:	Customer request					
EFFECT	N WEIGHT *		EFFE	CT ON BALAN	ICE *	
GUAR, WT. EMPTY	OPER. WT. EMPTY					
+4.0 lb	+4.0 lb			+1,055	INCH LB.	
EFFECT ON GUARANTEED F		ne				
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOS.  ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			alwaren	engine Ering	APPROVAL	
				ENGINEERING	AFTROVAL	
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE: RECURRING: NON-RECURRING: TOTAL:				
ACCEPTED.		CONVA	AIR, A DIVISION OF G	ENERAL DYNA	AICS CORP.	
BY:						
DATE:					orthography and a	

Japan Air Lines Change No. 9

CONVAIR: SD

Radio and Flight Director System, Interconnection of Title:

JAL/CONVAIR Meeting of 10 May 1960 Origin:

Reason for Change: Customer request

Description of Change:

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS:

Above illustration will be revised to show instrument switch for Nav. Glide Slope No. 1 and No. 2.

Page 46 - Change paragraph number of "PERISCOPIC SEXTANT (PROVISIONS ONLY)" from "3.14.2" to "3.14.2.1"

Add following over new Paragraph 3.14.2.1:

"3.14.2 FLIGHT AND NAVIGATIONAL INSTRUMENTS AND INTERCONNECTION: "

Add the following new paragraph after Paragraph 3.14.2.1:

INTERCONNECTION OF EQUIPMENT: The flight director "3.14.2.2 system shall be interconnected with VHF Navigation Systems No. 1 and No. 2, so that inputs to the flight director system may be selected from either VIIF navigation system.

Effect on Weight Empty: +4.0 pounds Effect on Balance:

+1,055 inch-pounds

Effect on Performance:

None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD-22-322	DATE:					
CUSTOMER: Japan Air Lines		MCL	MCL 15,067 DTD 10 May 1960			
CHANGE NO: 8		MODI	EL: 22-22	(Convair	"880")	
TITLE: Altitude tion of	Control, Sperry In	ntegrated	Flight S	ystem Ins	talla-	
ORIGIN: JAL/CONVAIR Meeting of 10 May 1960						
REASON FOR CHANGE: Cu	istomer request					
EFFECT	ON WEIGHT *		EFFE	CT ON BALAN	1CE *	
GUAR. WT. EMPTY	OPER, WT. EMPTY					
+4.0 lb	+4.0 lb			+1,326	INCH LB.	
EFFECT ON GUARANTEED P	ERFORMANCE: * None	е				
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	AL W	16/2/60				
				ENGINEERING	APPROVAL	
LATEST DATE OF ACCEPTA	NCE:	AIRPLANES	AFFECTED:			
SPECIAL PROVISIONS:		RECURRING	KING:	AIRPLANE:		
ACCEPTED:		CONVAIR, A	DIVISION OF G	ENERAL DYNA	WICS CORP.	
BY						
DATE:	1			a.	DN VAIR, 5. D. 6-12/7A	

Japan Air Lines CONVAIR: SD Change No. 8

Altitude Control, Sperry Integrated Flight System, Installa-Title:

tion of

JAL/CONVAIR Meeting of 10 May 1960 Origin:

Reason for Change: Customer request

Description of Change:

Page A-4, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Add the following item under "SPERRY FLIGHT NAVIGATION SYSTEM":

"1 Altitude Control

Sperry 620399-2"

Effect on Weight Empty: +4.0 pounds

Effect on Balance: +1,326 inch-pounds Effect on Performance: None

4 DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322	DATE:					
CUSTOMER: Japan Air Lines	MCL 15,065 DTD 10 May 1960					
CHANGE NO: 7	MODEL: 22-22 (Convair "880")					
TITLE: Flight Reference Selector Swoof	witch, and CoB Indicator, Installation					
ORIGIN: JAL/CONVAIR Meeting of 10 Ma	ıy 1960					
REASON FOR CHANGE: Customer request						
EFFECT ON WEIGHT *	EFFECT ON BALANCE *					
GUAR, WT. EMPTY OPER. WT. EMPTY +15.0 1b +15.0 1b	+4,250 INCH LB.					
EFFECT ON GUARANTEED PERFORMANCE: *  None						
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSE ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPO PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:	SAL Mayumy 16°					
LATEST DATE OF ACCEPTANCE:	AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:	EFFECT ON PRICE PER AIRPLANE; RECURRING: NON-RECURRING: TOTAL:					
ACCEPTED:	CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.					
BY:						
DATE-	*0. V 112.4 D. 6.1 17.4					

Title: Flight Reference Selector Switch, and C6B Indicator, Instal-

lation of

Origin: JAL/CONVAIR Meeting of 10 May 1960

Reason for Change: Customer request

Description of Change:

Page 44a, PILOT'S AND COPILOT'S INSTRUMENT PANELS:

Above illustration will be revised to show Flight Reference Selector Switch on center panel in lieu of Mode Selector Switch.

Page A-4, APPENDIX I-C, INSTRUMENTS AND RELATED EQUIPMENT:

Delete the below item under "SPERRY FLIGHT NAVIGATION SYSTEM":

"1 Mode Selector Sperry 1778905-2"

Add the following item under "SPERRY FLIGHT NAVIGATION SYSTEM"

"1 Flight Reference Selector Sperry 620361-601"

Revise the below items under "SPERRY C-11 REMOTE COMPASS SYSTEM" as follows:

From:	2 2 2	Indicator, Radio Magnetic tion (with Power Failure F Rack Amplifier, Servo	Direc- lag) Sperry Sperry Sperry	1777213-623 614937-10 618379"
To:	"2	Indicator, Radio Magnetic tion (with Power Failure F	lag)	
	248	(C-6B) Rack, Compass Amplifier, Servo Amplifier, Servo	Sperry Sperry Sperry Sperry	614937-11 618379 614937-51"

Effect on Weight Empty: +15.0 pounds

Effect on Balance: +4,250 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIE GO CALIFORNA

SPEC NO.: ZD- 22-322			DATE:			
CUSTOMER: Japan Air Lines			MCL 15,064 DTD 9 May 1960			
CHANGE NO. 6			MODEL: 22-22 (Convair "880")			
TITLE Additional Portable Oxygen Bottles, Installation of						
ORIGIN JAL/CONVAI	R Meeting of 26 Ap	oril l	1960			
REASON FOR CHANGE: Customer request						
	N WEIGHT *		EFFECT ON BALANCE *			
	OPER, WT. EMPTY					
None	+26.0 lb		+18,607 INCH LB.			
EFFECT ON GUARANTEED P	ERFORMANCE: * None					
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSA ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			ENGINEERING APPROVAL			
LATEST DATE OF ACCEPTANCE:			ANES AFFECTED:			
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:				
ACCEPTED-		CONVA	AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:						
DATE:						

Title: Additional Portable Oxygen Bottles, Installation of

Origin: JAL/CONVAIR Meeting of 26 April 1960

Reason for Change: Customer request

Description of Change:

Page 94, Paragraph 3.19.5.6 PORTABLE BOTTLES:

Revise the second sentence as follows:

"Three 310-liter portable oxygen bottles shall be provided, From: two located in the aft entrance area and one located in the forward coat compartment."

"Six, seven-cubic foot portable oxygen bottles shall be provided, three located in the aft L.H. coat closet, two To: the forward R.H. coat closet and one in the forward L.H. coat closet."

### Page A-15, APPENDIX I-C, OXYGEN EQUIPMENT:

Revise the below items as follows:

\*Bottle, Portable Oxygen From: (7-cu ft) Scott 5500-B1A-BF20B Mask, Oxygen 3 To: \*Bottle, Portable Oxygen (7-cu ft) Scott 5500-B1A-BF20B 6 Mask, Oxygen (A8B) Scott 23275

Effect on Operating Weight Empty: +26.0 pounds

Effect on Weight Empty:

Effect on Balance:

Effect on Performance:

None

+18,607 inch-pounds

None

4 DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322		DATE:						
CUSTOMER: Japan Air Lines			MCL 15,063 DTD 9 May 1960					
CHANGE NO: 5			MODEL: 22-2 (Convair "880")					
	TITLE: Toilet Paper Dispenser and R	lazor	Blade Disposal, Installation of					
	ORIGIN: JAL/CONVAIR Meeting of 26 April 1960							
	REASON FOR CHANGE: Customer request							
	EFFECT ON WEIGHT *		EFFECT ON BALANCE *					
	GUAR, WT. EMPTY OPER, WT. EMPTY							
	+5.0 lb +5.0 lb		+4,200 INCH LE					
	EFFECT ON GUARANTEED PERFORMANCE: *  None							
NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOSAL ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPON PRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			MW44166 ENGINEERING APPROVAL					
LATEST DATE OF ACCEPTANCE:		AIRPL	ANES AFFECTED:					
SPECIAL PROVISIONS:		EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:						
ACCEPTED: C		CONVA	IR, A DIVISION OF GENERAL DYNAMICS CORP.					
	RY.							
DATE:			70 4 14 July 9-1417 A					
	The state of the s		7 7 7 1 1 1 1 1 4					

CONVAIR: SD

14 73

> Toilet Paper Dispenser and Razor Blade Disposal, Installa-Title:

tion of

JAL/CONVAIR Meeting of 26 April 1960 Origin:

Reason for Change: Customer request

Description of Change:

Page 87, Paragraph 3.19.2.2.3 MISCELLANEOUS LAVATORY EQUIPMENT:

Add the following items to the list of equipment to be installed in the lavatory:

"One toilet paper dispenser (interlock leaf) One razor blade disposal container"

Effect on Weight Empty: +5.0 pounds Effect on Balance:

+4,200 inch-pounds

Effect on Performance: None

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIEGO, CALIFORNIA

SPEC NO.: ZD- 22-322		DATE:			
CUSTOMER: Japan Air Line	MCL 15,056 DTD Revised 5 May 1960				
CHANGE NO: 4	MODEL: 22-22 (Convair "880")				
TITLE: Overhead Convenience Pods, Installation of; and Clarification of Seat Spacing					
ORIGIN: JAL/CONVAIR Meeting of 26 April 1960					
REASON FOR CHANGE: Customer request .					
EFFECT ON WEIG	SHT *	EFFECT ON BALANCE *			
	+96.0 lb	+60,960 INCH LB.			
EFFECT ON GUARANTEED PERFOR	RMANCE: * None				
* NEGLIGIBLE CHANGES WILL BE A TOTALS REFLECTED IN A FUTUR ACCEPTANCE OF THIS CHANGE IS PRIOR ACCEPTANCE OF THE FOL	S DEPENDENT UPON	alwayuung 60  ENGINEERING APPROVAL			
LATEST 'DATE OF ACCEPTANCE:	AIRPL	ANES AFFECTED:			
RE NO		CT ON PRICE PER AIRPLANE:  RRING: ECURRING: -:			
ACCEPTED: CON		AIR, A DIVISION OF GENERAL DYNAMICS CORP.			
BY:					
DATE		TONVAIH,5,0, 6:12174			

CONVAIR: SD Japan Air Lines Change No. 4

Title: Overhead Convenience Pods, Installation of; and Clarification

of Seat Spacing

Origin: JAL/CONVAIR Meeting of 26 April 1960

Reason for Change: Customer request

Description of Change:

This proposal has no effect on specification language, and provides new spacer panels in overhead convenience pods to permit pod adjustment from 31 to 37 inches, with the first class seats at 42-inch spacing and the coach seats at 34-inch spacing as presently specified in the specification. The convenience pods will be installed at 34-inch spacing starting at ten inches forward of Station 1217 (both R.H. and L.H. sides) extending forward to Station 446. This adds eight additional pods over original configuration. It is not intended that 31-inch spacing will be possible throughout the cabin, but only at a few locations at any one time.

This proposal also installs two additional passenger public address speakers, one on each side of the cabin.

Effect on Weight Empty: Effect on Balance: Effect on Performance:

+96.0 pounds +60,960 inch-pounds None

A DIVISEN OF SENERAL DYNAMICS CORPORATION.
SATI DIVIDO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:					
CUSTOMER. Japan Air Lines			MCL _	15,054	DTD 3	evis May	<b>ed</b> 196	
CHANGE NO: 3			MODEL	22-22	(Conva	ir "	880	
TITLE Forward R Installat	ight Hand Coat Cor ion of	mpartm	ment a	and First	Class	Curt	ain	
ORIGIN JAL/Conva	ir meeting, 26 Ap:	ril l	960					
REASON FOR CHANGE:	Customer request							
EFFECT ON WEIGHT '			EFFECT ON BALANCE					
SUAR, WT. EMPTY	OPER. WT. EMPTY							
-57.0 lb	-57.0 lb	-24,800 INC						
EFFECT ON GUARANTEED P	ERFORMANCE * None							
* NEGLIGIBLE CHANGES WILL BE ACCUMULATED AND TOTALS REFLECTED IN A FUTURE CHANGE PROPOS ACCEPTANCE OF THIS CHANGE IS DEPENDENT UPOPPRIOR ACCEPTANCE OF THE FOLLOWING CHANGES:			alw f	guenn J.T. Plund	SINEERING	g app	ROVA	
LATEST DATE OF ACCEPTANCE:			AIRPLANES AFFECTED:					
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:					
ACCEPTED		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.						
BY:								
DATE:								

CONVAIR: SD

Japan Air Lin Change No. 3

Page 1 of 2

Title: Forward Right Hand Coat Compartment and First Class Curtain

Installation of

Origin: JAL/Convair meeting, 26 April 1960

Reason for Change: Customer request

Description of Change:

Page 84, Paragraph 3.19.2.1.1 BUFFETS:

Revise the paragraph to read as follows:

"Five Buyer-furnished buffet units shall be installed. Units No. 1 and No. 2 (two-section unit) shall be located in the forward right-hand area. Units No. 3 and No. 4 shall be located in the right-hand service door area. Unit No. 5 shall be located forward of the aft entrance door on the left-hand side."

Page 87, Paragraph 3.19.2.3 COAT STOWAGE:

Specify a third fixed coat compartment. The paragraph will ther read as follows:

"Three fixed passenger coat stowage compartments shall be install and shall be located as follows: One forward of the forward may entrance door on the left side, one aft of the aft main entrance door, and one aft of the Number 2 buffet on the right side. The installation shall be as shown on the illustration herein (See also 3.19.3.1.2)."

Page 89, Paragraph 3.19.3.1.2 MOVABLE COAT COMPARTMENTS:

In the second sentence, delete reference to a third track mounte coat compartment. The sentence will then read as follows:

"Both coat compartments shall utilize the passenger seat tracks permit fore and aft movement at one-inch increments, except in tareas of the overhead life raft stowage and the emergency exit locations."

CONVAIR: SD

Japan Air Ling Change No. 3

Page 2 of 2

Page 90, Paragraph 3.19.3.5 CURTAINS:

Add the following as the next-to-the-last sentence:

"An aisle curtain shall be provided at the forward end of the first-class section."

Page A-15, APPENDIX I-C, FURNISHINGS

### Delete:

"1 Coat Compartment, R.H. 23.6-Inch Movable"

Effect on Weight Empty: -57.0 pounds

Effect on Balance: -24,800 inch-pounds

Effect on Performance: None

The following will not appear in the specification language:

The window at Station 441.8 will be deleted and structural provisions only for a window shall be incorporated.

A LIVERON OF GENERAL DYNAMICS CORROBATION SAN DIE 60, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:						
CUSTOMER: Japan A	ir Lines		MCL_	15,053	DTD	Revised 3 May 19			
CHANGE NO 2						onvair "8			
TITLE Trilingua	l and Bilingual S:	igns,	Insta:	llation c	f Ad	ditional			
ORIGIN: JAL/Conva	ir Meeting, 26 Apr	ril 19	60						
REASON FOR CHANGE: C	ustomer request								
FFFECT ON WEIGHT			EFFECT ON BALANCE						
GUAR, WT. EMPTY	OPER, WT. EMPTY								
Neg. Neg.				Neg		INCH			
EFFECT ON GUARANTEED F		Hone							
* NEGLIGIBLE CHANGES WILL TOTALS REFLECTED IN A ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF T	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N			GINEER	RING APPROV			
LATEST DATE OF ACCEPTANCE:			AIRPLANES AFFECTED:						
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE  RECURRING: NON-RECURRING: TOTAL:						
ACCEPTED:		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP							
ВҮ									
DATE:									

A DIVISION OF GENERAL DYNAMICS CORPORATION SAN DIFGO, CALIFORNIA

SPEC NO.: ZD- 22-322			DATE:					
CUSTOMER: Japan Air Lines			MCL 15,057 DTD Revised 3 May 1960					
CHANGE NO. 1			MODEL: 2					
TITLE: Tourist	Class Curtain, In	stalla	ation of					
ORIGIN: JAL/Conv	air meeting, 26 A	pril ]	1960			4		
REASON FOR CHANGE:	Customer request			,				
EFFECT ON WEIGHT *			EFFECT ON BALANCE *					
GUAR. WT. EMPTY	AR. WT. EMPTY OPER, WT. EMPTY							
+8.0 lb		+9,696 INCH LB.						
EFFECT ON GUARANTEED P	ERFORMANCE: *							
	None							
ACCEPTANCE OF THIS CHAPRIOR ACCEPTANCE OF THE	FUTURE CHANGE PROPOS ANGE IS DEPENDENT UPO	SAL N	alique			\PPROVAL		
LATEST DATE OF ACCEPTA	NCE;	AIRPL	ANES AFFECT	ED:				
SPECIAL PROVISIONS:			EFFECT ON PRICE PER AIRPLANE:  RECURRING:  NON-RECURRING:  TOTAL:					
ACCEPTED		CONVAIR, A DIVISION OF GENERAL DYNAMICS CORP.						
8Y:								
DATE:					164	in a succession of		

CONVAIR: SD Japan Air Lines Change No. 1

Title: Tourist Class Curtain, Installation of

Origin: JAL/Convair meeting, 26 April 1960

Reason for Change: Customer request

Description of Change:

Page 90, Paragraph 3.19.3.5 CURTAINS:

Add the following after the third sentence:

"A curtain, tied to left and the right hand sides, shall be provided at the aft end of the tourist section."

Effect on Weight Empty: Effect on Balance:

+8.0 pounds +9,696 inch-pounds

Effect on Performance:

None